2 In plane loading – walls and beams

2.3 Compatibility and deformation capacity

Learning objectives

Within this chapter, the students are able to:

- determine the behaviour of concrete as a function of the compressive strength and the cracking state.
- recognise the assumption of limit analysis methods for the materials having sufficient deformation capacity to reach the plastic solution without rupturing, and the existence of approaches to verify the deformation capacity of the materials.
- evaluate plastic redistributions of internal forces in hyperstatic systems (beams and frames) subjected either to external loads or imposed deformations, and calculate the deformation demand in elements subjected to bending or normal actions.
- estimate the deformation capacity of a structure subjected to bending or normal actions.
 - explain the tension-stiffening effect and how it affects the structural behaviour.
 - illustrate the main assumptions and behaviour of bonded reinforcement according to the Tension Chord Model.

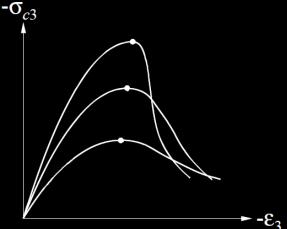
2 In plane loading – walls and beams

2.3 Compatibility and deformation capacity

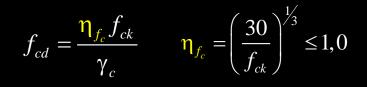
A) Behaviour of concrete in compression

Main factors influencing the equivalent strength to be considered in plastic calculations

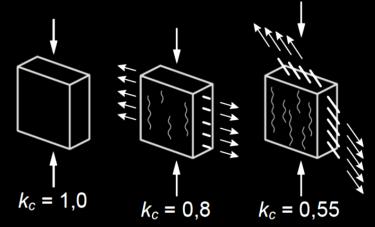
Strain softening after peak strength (material effect)



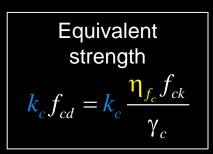
The concrete brittleness (i.e. relative amount of softening) increases with the compressive strength and also the reduction of the strength to be accounted for (η_{fc}) .



Influence of transverse cracking on concrete strength (structural effect)



Reduction factor to account for this effect (k_c) can be determined in a more refined manner based on the state of deformations (see following slides).



Dependence of the concrete compressive strength and shear resistance on the strain state

Tests have shown that the compressive strength in membrane elements is reduced by (imposed) transverse strains.

In 1986, Vecchio and Collins proposed reducing the compressive strength by a factor $1/(0.8+170\cdot\epsilon_1) \le 1$ (assuming "average" concrete stresses). This also takes implicitly other effects into account.

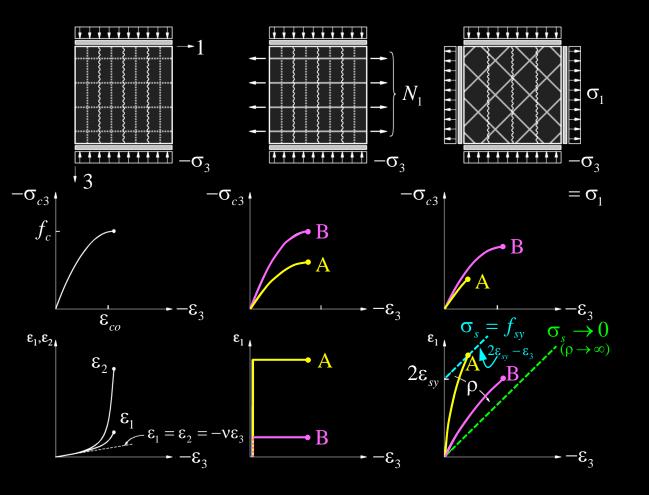
In 1998, Kaufmann proposed to consider additionally the (already known) inversely proportional increase of the compressive strength with the cylinder compressive strength:

 $f_{ce} = \frac{f_{c,cyl}^{2/3}}{0.4 + 30 \cdot \varepsilon_1}$

On the basis of this and other work, SIA 262 has introduced the following coefficient for the verification of webs of beams:

 $k_{c} = \frac{1}{1, 2 + 55\varepsilon_{1}} \le 0.65$

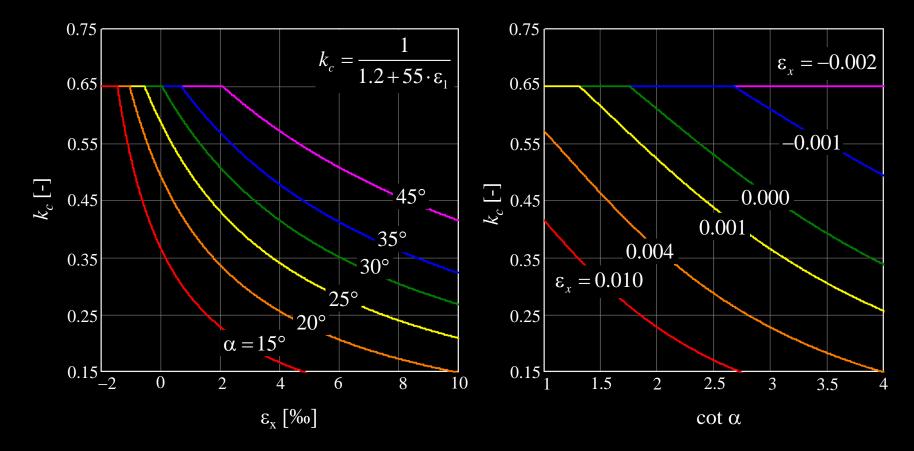
This can be applied in a more general way to any structural member when removing the 0.65 upper-bound



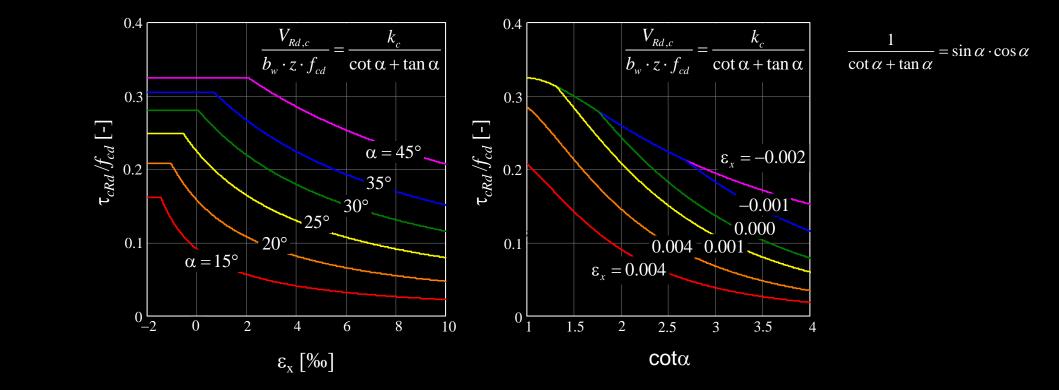
 $k_{c}(\varepsilon_{x},\alpha) = \frac{1}{1.2 + 55 \cdot \varepsilon_{1}} \le 0.65 \qquad \varepsilon_{1} = \varepsilon_{x} + (\varepsilon_{x} + 0.002) \cdot \cot^{2} \alpha \quad \blacktriangleleft$ 70 r $\alpha = 15^{\circ}$ Assumption $\varepsilon_3 = -0.002$ $\gamma/2$ 20° Х 50 25° 30° ³⁰ 30 35° $(\varepsilon_x - \varepsilon_3) \cot^2 \alpha$ 45° 3 1 E α $(\varepsilon_x - \varepsilon_3) \cot \alpha$ α 10 $\varepsilon_1 = \varepsilon_x + (\varepsilon_x + 0.002) \cdot \cot^2 \alpha$ -10Q 10 -20 2 6 8 ε_x [‰]

Concrete compressive strength and shear resistance as a function of the strain state

Concrete compressive strength and shear resistance as a function of the strain state



 $\rightarrow k_c \cdot f_c$ is largely reduced for flat inclinations of the compression field and for plastic strains of the tension chord



Concrete compressive strength and shear resistance as a function of the strain state

- $\rightarrow k_c f_c$ is largely reduced for flat inclinations of the compression field and for plastic strains of the tension chord
- → Concrete compressive stresses increase sharply with flat inclinations (see above)
- \rightarrow Very flat inclinations do not make sense when dimensioning, but are often necessary when assessing old bridges
- \rightarrow Attention to plastic internal force redistributions from the support (= large shear force)

2 In plane loading – walls and beams

2.3 Compatibility and deformation capacity

B) Behaviour of bonded reinforcement

 \rightarrow ODE of 2nd order

Differential equations of bond

General bond-slip law

Equilibrium of an element with length dx:

 $\frac{d\sigma_c}{dx} = -\frac{\varnothing \pi \tau_b + q_x}{A_c (1 - \rho)}; \qquad \qquad \frac{d\sigma_s}{dx} = \frac{4\tau_b}{\varnothing} \qquad \rightarrow \text{ODE of } 1^{\text{st}} \text{ order}$

Considering linear elastic material behaviour:

 $\frac{d^2\delta}{dx^2} = \frac{4\tau_b}{\emptyset E_s} + \frac{\emptyset \pi \tau_b + q_x}{A_c E_c (1 - \rho)}$

Simplified bond-slip law, used in TCM

Equilibrium of an element with length dx:

 $\frac{d\sigma_c}{dx} = -\frac{\varnothing \pi \tau_b + q_x}{A_c (1 - \rho)} = \text{const}^*; \qquad \frac{d\sigma_s}{dx} = \frac{4\tau_b}{\emptyset} = \text{const}^* \longrightarrow \text{linear (integrate once)}$

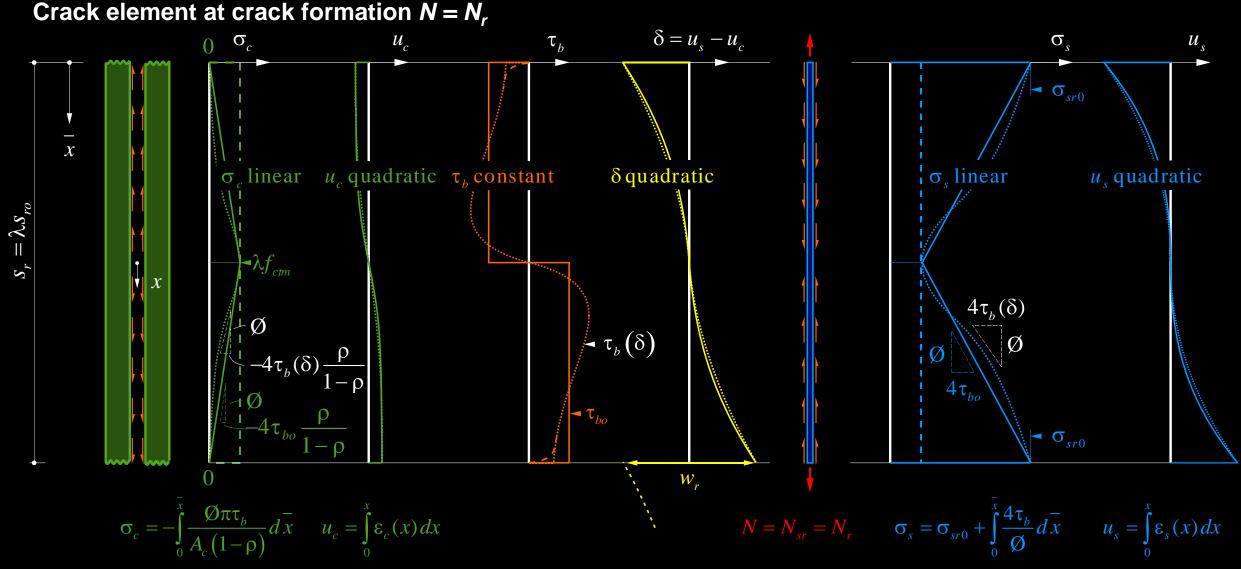
Considering linear elastic material behaviour:

$$\frac{d^2\delta}{dx^2} = \frac{4\tau_b}{\emptyset E_s} + \frac{\emptyset \pi \tau_b + q_x}{A_c E_c (1 - \rho)} = \text{const}^* \quad \rightarrow \text{quadratic (integrate twice)}$$

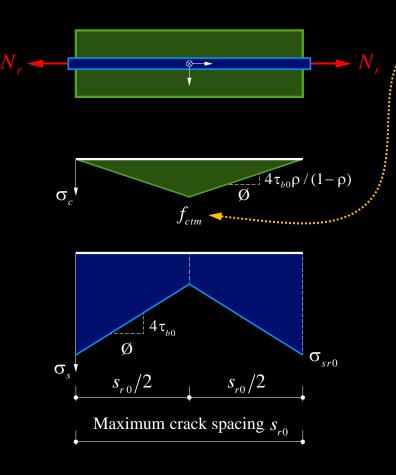
$$* \text{ if } q_x = \text{const}$$

dxU. δ U $\sigma_c + d\sigma_c$ $\sigma_{c} + d\sigma_{c}$ х $\tau_{b0} = 2f_{ctm}$ ► δ $\delta_{v}\left(\sigma_{s}=f_{sv}\right)$

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View of a tension chord (total cross-section A_c), reinforced with bar with diameter Ø ([6], page 3.5f)



Concrete stress in the middle of the element with length s_{r0} (maximum \cdot crack spacing) is $\sigma_c = f_{ctm}$ i.e. another crack could form there.

$$S_{rm0} \approx \frac{\emptyset}{4} \cdot \left(\frac{1}{\rho_t} - 1\right)$$

Thus the minimum crack spacing is:

$$s_{r,min} = s_{r0}/2$$

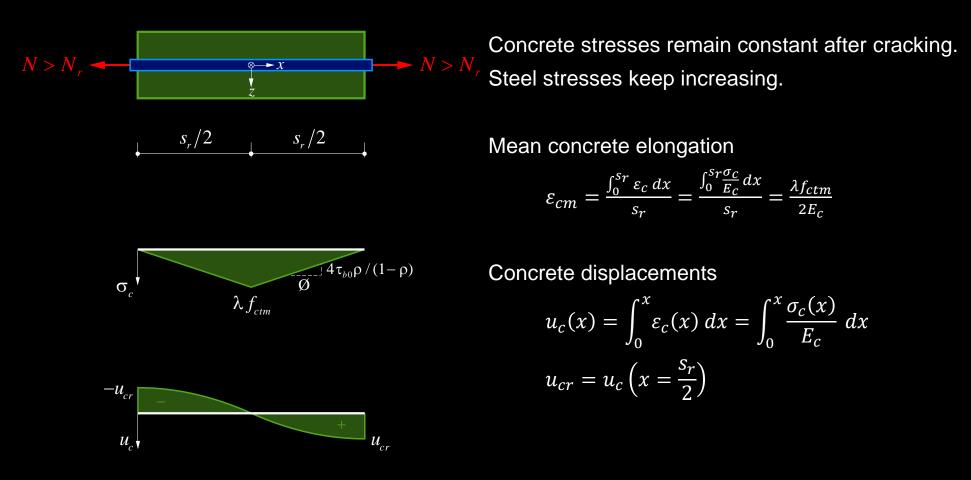
Generally, the crack spacing varies with parameter λ :

$$s_r = \lambda \cdot s_{r0} \qquad \left(\frac{1}{2} < \lambda < 1\right)$$

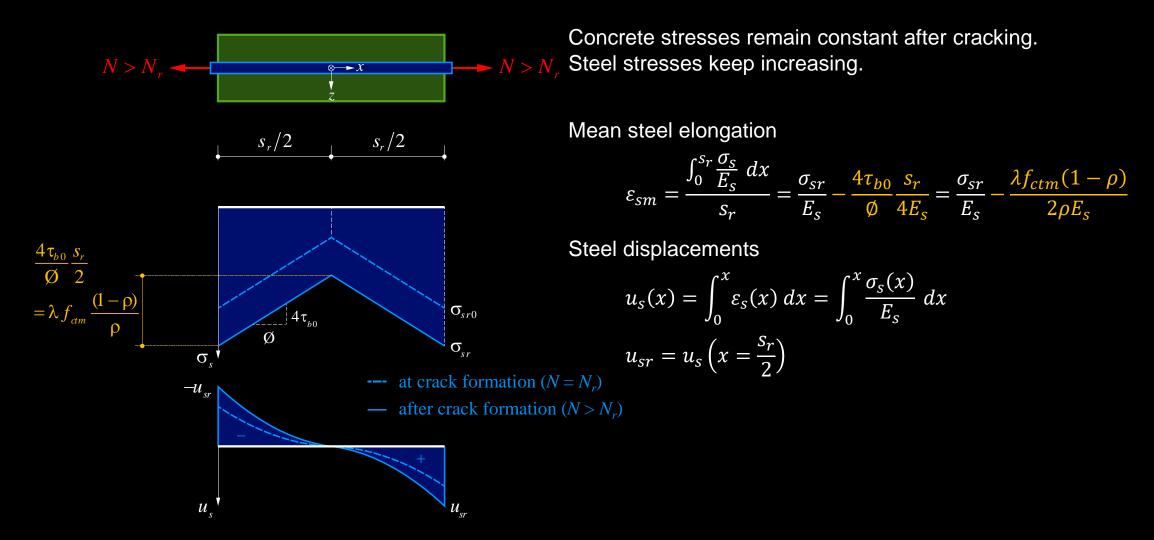
→ Theoretical limits of the crack spacing with fully developed crack pattern!

SN: If the cracks form because of applied loads, the fully developed crack pattern forms at once (theoretically).

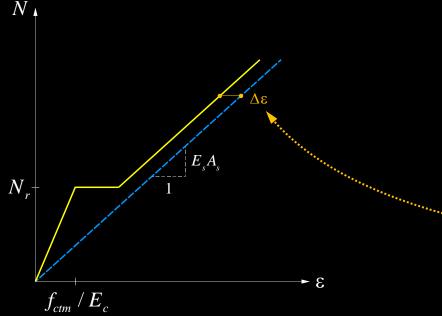
Increase of normal force after crack formation $N > N_r$ ([6], page 3.5f)



Increase of normal force after crack formation $N > N_r$ ([6], page 3.5f)



Increase of normal force after crack formation $N > N_r$ ([6], page 3.5f)



N- ϵ - and σ_{sr} - ϵ -diagrams : Reduction of the elongation of the bare steel by $\Delta\epsilon$ ($\Delta\epsilon$ remains constant until yielding).

NB: Good approximation for w_r (small ρ)

$$\frac{\emptyset/4\rho}{2E_s} \left(\frac{N}{A_s} - \frac{f_{ctm}}{4\rho}\right) \le w_r \le \frac{\emptyset/4\rho}{E_s} \left(\frac{N}{A_s} - \frac{f_{ctm}}{2\rho}\right)$$

Concrete stresses remain constant after cracking. Steel stresses keep increasing.

Average concrete elongation

 $\varepsilon_{cm} = \lambda f_{ctm} / (2E_c)$

Steel elongation at crack

$$\varepsilon_{sr} = \sigma_{sr}/E$$

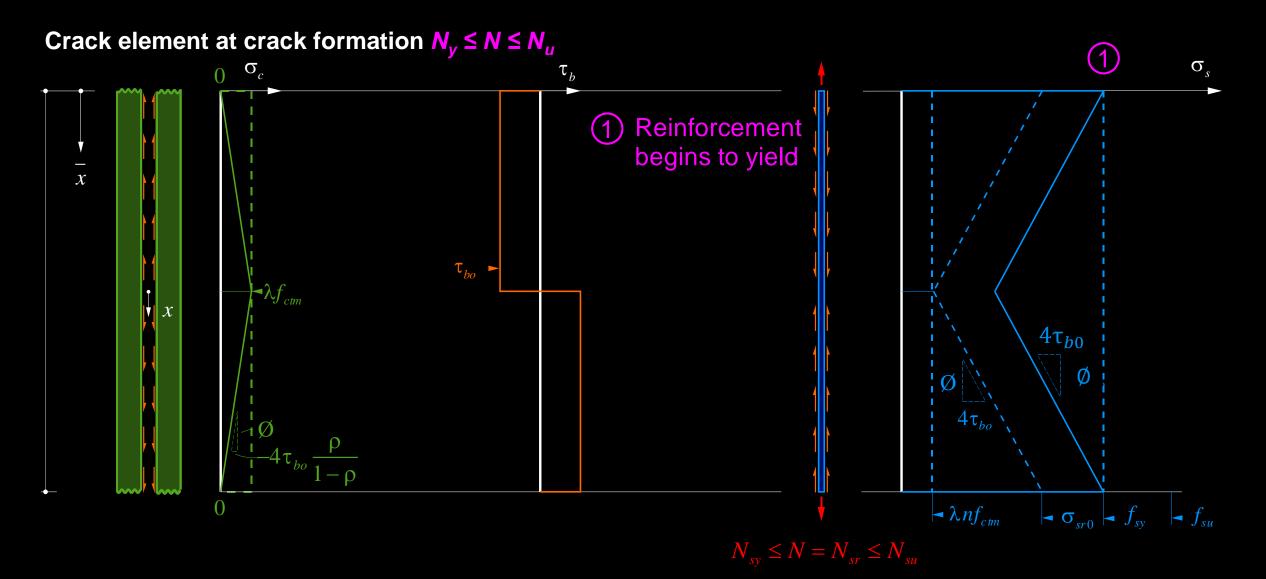
Mean steel elongation

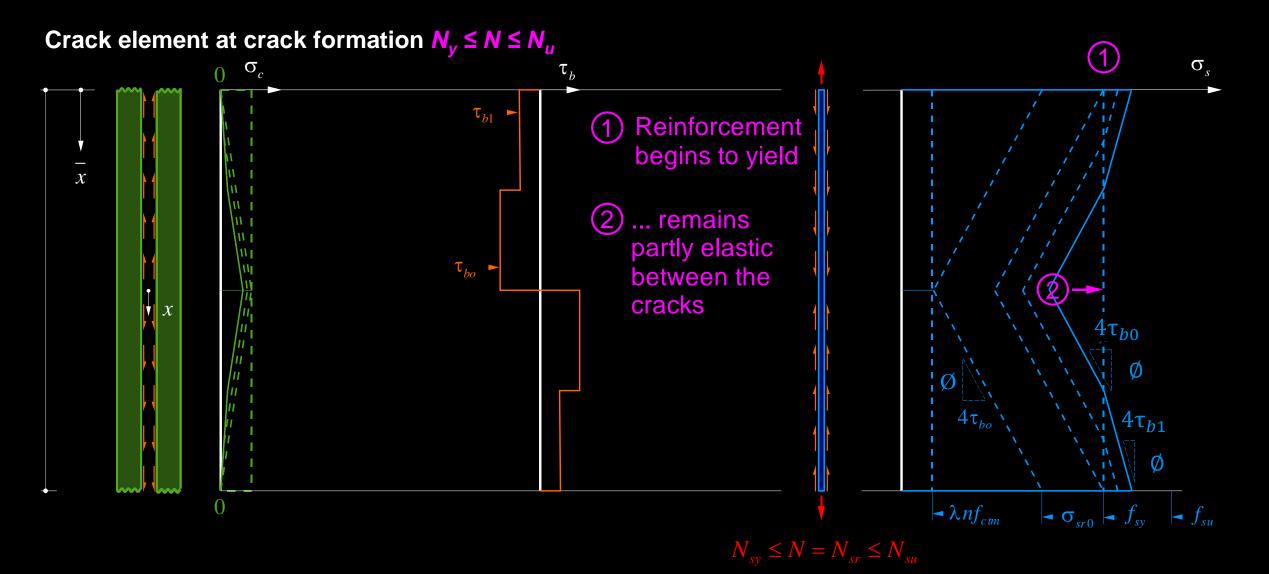
$$\varepsilon_{sm} = \frac{\sigma_{sr}}{E_s} - \frac{\tau_{b0}}{\emptyset} \frac{s_r}{E_s} = \frac{\sigma_{sr}}{E_s} - \frac{\lambda f_{ctm} (1 - \rho)}{2\rho E_s}$$

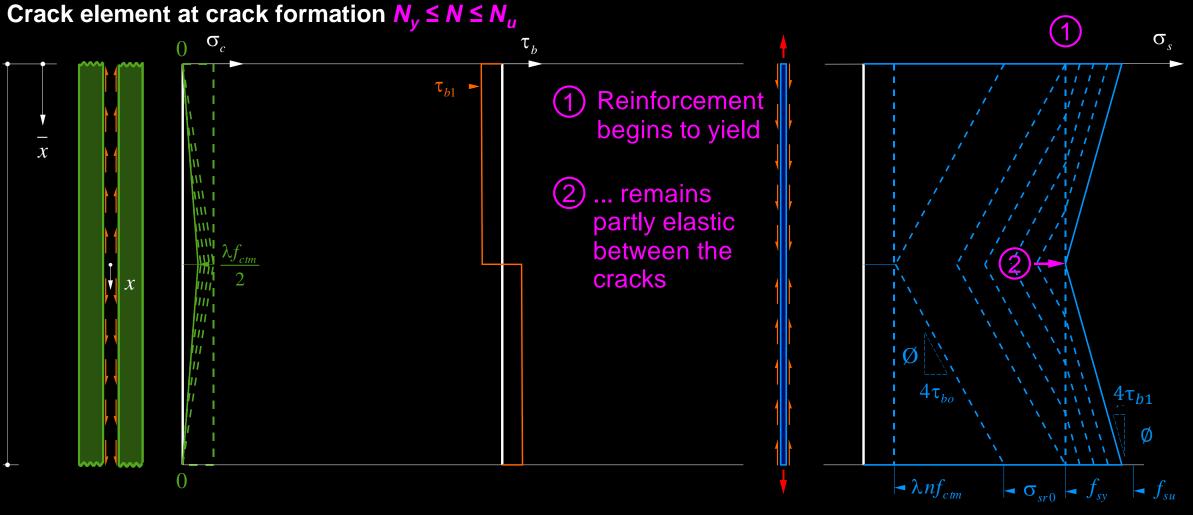
Crack widths: Difference of the mean steel and concrete strains multiplied by $s_r (\lambda = 0.5...1)$:

$$w_r = s_r \left[\frac{\sigma_{sr}}{E_s} - \frac{\lambda f_{ctm} (1 - \rho)}{2\rho E_s} - \frac{\lambda f_{ctm}}{2E_c} \right] = \frac{\lambda s_{r0} (2\sigma_{sr} - \lambda \sigma_{sr0})}{2E_s}$$

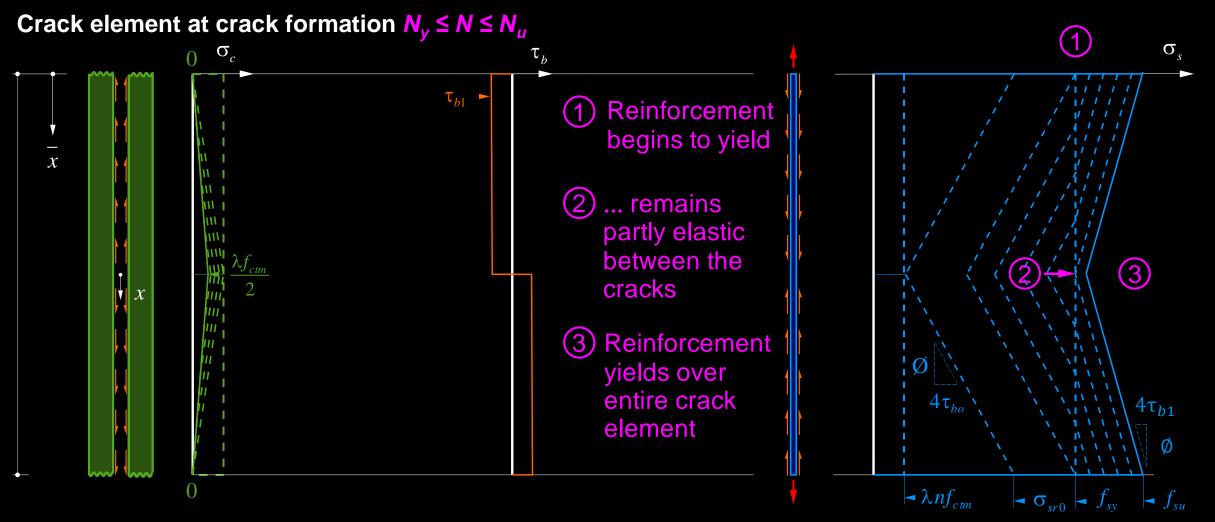
with $\sigma_{sr} = N/A_s$
$$\frac{s_{r0}}{2E_s} \left(\sigma_{sr} - \frac{\sigma_{sr0}}{4} \right) \le w_r \le \frac{s_{r0}}{E_s} \left(\sigma_{sr} - \frac{\sigma_{sr0}}{2} \right)$$







 $N_{sy} \leq N = N_{sr} \leq N_{sr}$



 $N_{sy} \leq N = N_{sr} \leq N_{sr}$

Closed form solution for a bilinear steel stress-strain relationship

① Elastic reinforcement over entire crack element

$$\sigma_{sr} \leq f_{sy}$$

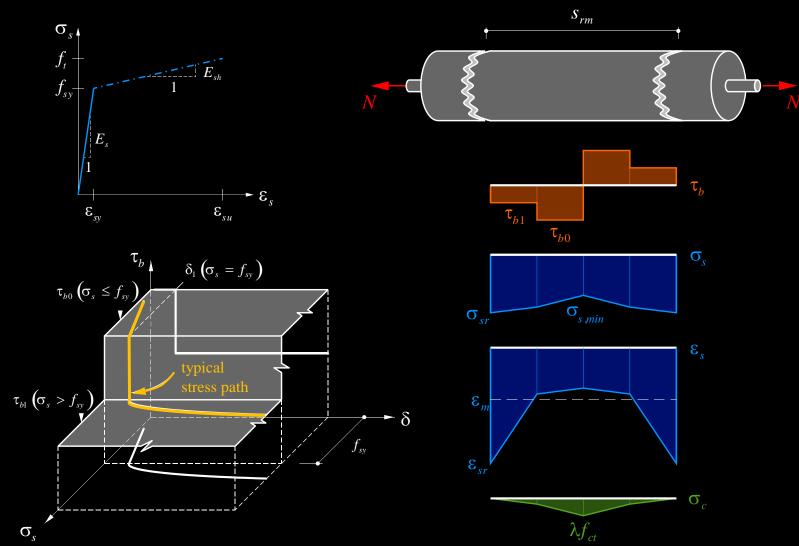
 Reinforcement yields near cracks, elastic between cracks

 $f_{sy} \leq \sigma_{sr} \leq \left(f_{sy} + \frac{2\tau_{b1}s_r}{\varnothing}\right)$

③ Reinforcement yields over entire crack element

$$\left(f_{sy} + \frac{2\tau_{b1}s_r}{\varnothing}\right) \le \sigma_{sr} \le f_{su}$$

$$\begin{aligned} f_{sr} &= E_s \varepsilon_{sm} + \frac{\tau_{b0} s_r}{\varnothing} \\ f_{sr} &= E_s \varepsilon_{sm} + \frac{\tau_{b0} s_r}{\varnothing} \\ f_{sm} &= \frac{\sigma_{sr}}{E_s} - \frac{\tau_{b0} s_r}{E_s \varnothing} = \frac{\sigma_{sr}}{E_s} - \lambda \frac{f_{ctm} (1 - \rho)}{2E_s \rho} \\ & \Delta \varepsilon_0 = \frac{\tau_{b0} s_r}{E_s \varnothing} = \lambda \frac{f_{ctm} (1 - \rho)}{2E_s \rho} \\ f_{sr} &= f_{sy} + 2 \frac{\frac{\tau_{b0} s_r}{\varnothing} - \sqrt{\left(f_{sy} - E_s \varepsilon_{sm}\right) \frac{\tau_{b1} s_r}{\varnothing} \left(\frac{\tau_{b0}}{\tau_{b1}} - \frac{E_s}{E_{sh}}\right) + \frac{E_s}{E_{sh}} \tau_{b0} \tau_{b1} \frac{s_r^2}{\varnothing^2}}{\left(\frac{\tau_{b0}}{\tau_{b1}} - \frac{E_s}{E_{sh}}\right)} \\ f_{sm} &= \frac{\left(\sigma_{sr} - f_{sy}\right)^2 \varnothing}{4E_{sh} \tau_{b1} s_r} \left(1 - \frac{E_{sh} \tau_{b0}}{E_s \tau_{b1}}\right) + \frac{\left(\sigma_{sr} - f_{sy}\right) \tau_{b0}}{E_s} + \left(\varepsilon_{sy} - \frac{\tau_{b0} s_r}{E_s \varnothing}\right) \\ f_{sm} &= f_{sm} + E_s \left(\varepsilon_{sm} - \frac{f_{sy}}{E_s \tau_{b1}}\right) + \frac{\tau_{b1} s_r}{E_s} \\ & \text{where steel} \quad A \varepsilon \end{aligned}$$



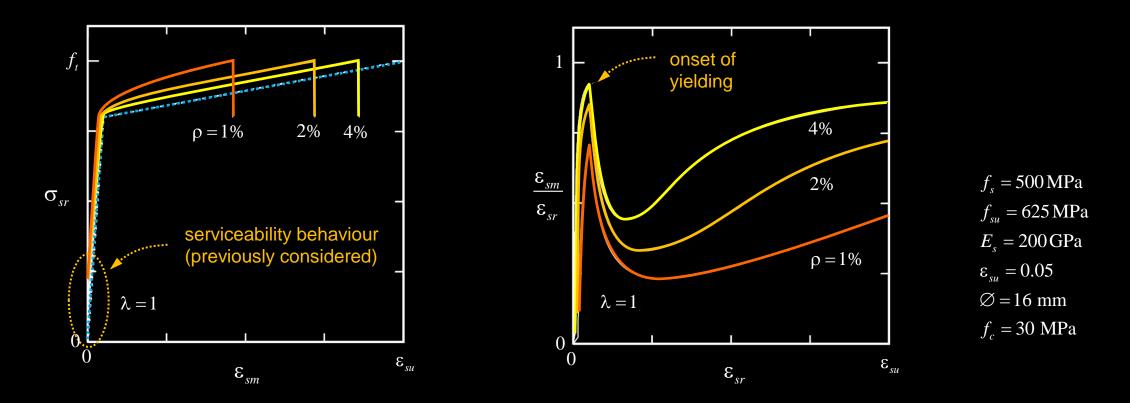
Constitutive relationship of the bonded reinforcement (tension chord model with bilinear bare reinforcement):

Load-deformation behaviour considering bond (influence at high loads)

- \rightarrow No influence on tensile resistance
- $\rightarrow\,$ Stiffer behaviour than bare steel

Ratio of average elongation to maximum elongation at the cracks considering bond

- \rightarrow Heavy drop after onset of yielding
- \rightarrow Pronounced influence on ductility!

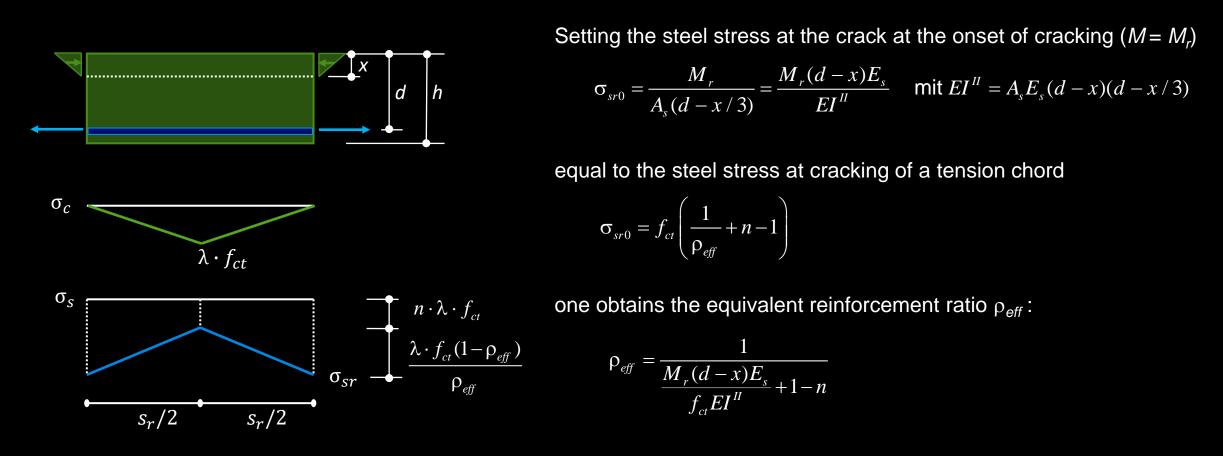


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Behaviour of the bonded reinforcement

Application to loading cases different than uniaxial tension

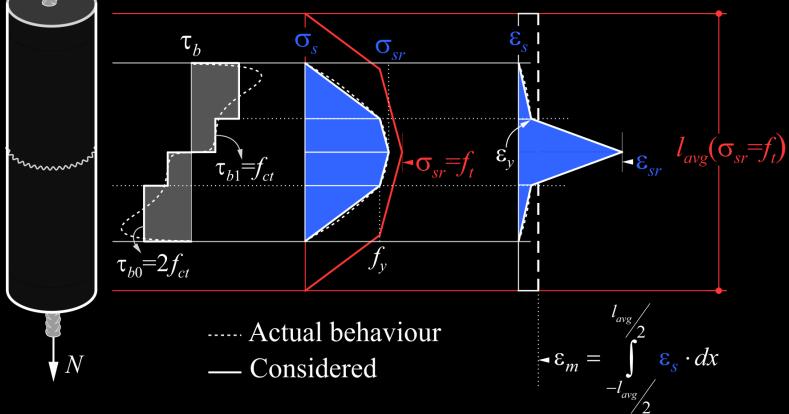
Simple bending (SB I): Elastic bending stiffness – tensile stiffness [6], page 2.16f



Behaviour of the bonded reinforcement

Tension stiffening for non-stabilised crack patterns (pull-out model)

for $\rho < \rho_{cr} \approx 0.6\% \rightarrow$ Reinforcement is NOT able to carry the cracking load without yielding and the tension chord model is not applicable. Cracks are controlled by other reinforcement and a stabilized crack pattern is not generated.

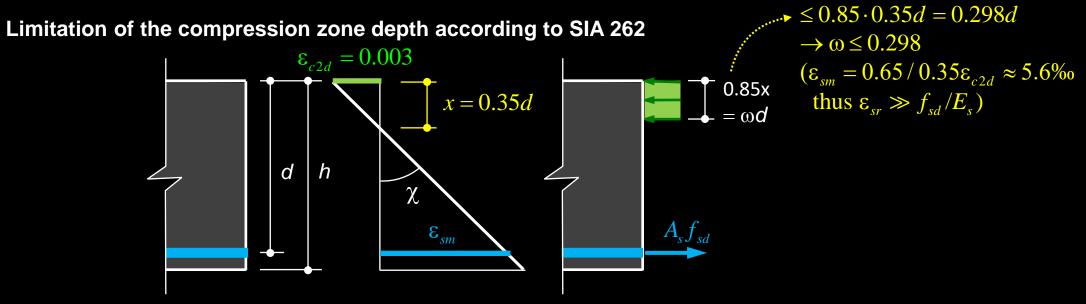


- A pull-out tension stiffening model can be formulated for these situations by assuming (a) independent cracks and (b) the same bond slip model as for the tension chord model.
- A certain crack spacing (*I_{avg}*) should be assumed to compute the average reinforcement strain.

2 In plane loading – walls and beams

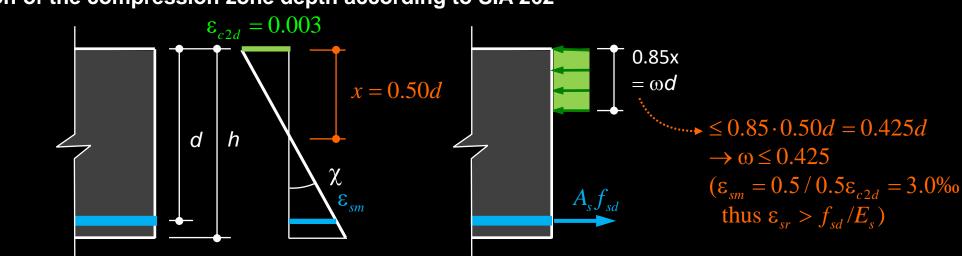
2.3 Compatibility and deformation capacity

C) Deformation capacity of beams



Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

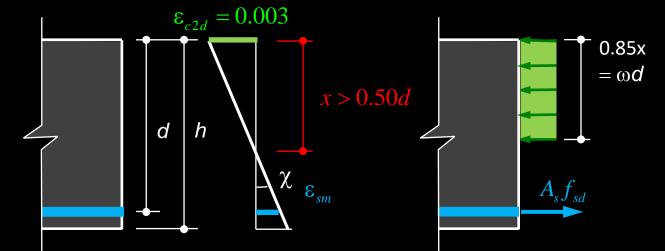
• x/d ≤ 0.35: Internal force redistributions without verification of deformation capacity $x/d \le 0.35 \rightarrow \omega \le 0.298 \rightarrow M_{Rd} \le bd^2 f_{cd} \omega \cdot (1 - \omega/2) = 0.253 \cdot bd^2 f_{cd}$



Limitation of the compression zone depth according to SIA 262

Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

• 0.35 \leq x/d \leq 0.5: Internal force redistributions with verification of deformation capacity $x/d \leq 0.50 \rightarrow \omega \leq 0.425 \rightarrow M_{_{Rd}} \leq bd^2 f_{_{cd}} \omega \cdot (1 - \omega/2) = 0.335 \cdot bd^2 f_{_{cd}}$

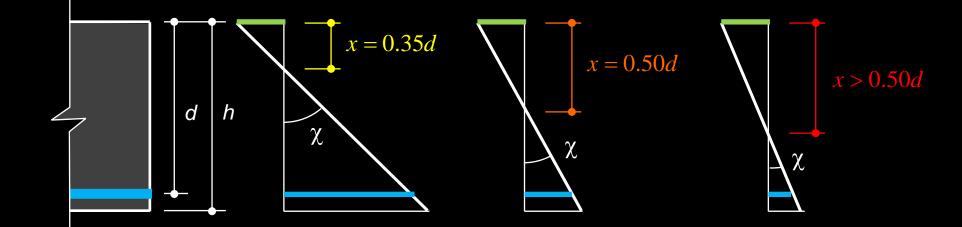


Limitation of the compression zone depth according to SIA 262

Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

• x/d > 0.50: is to be avoided

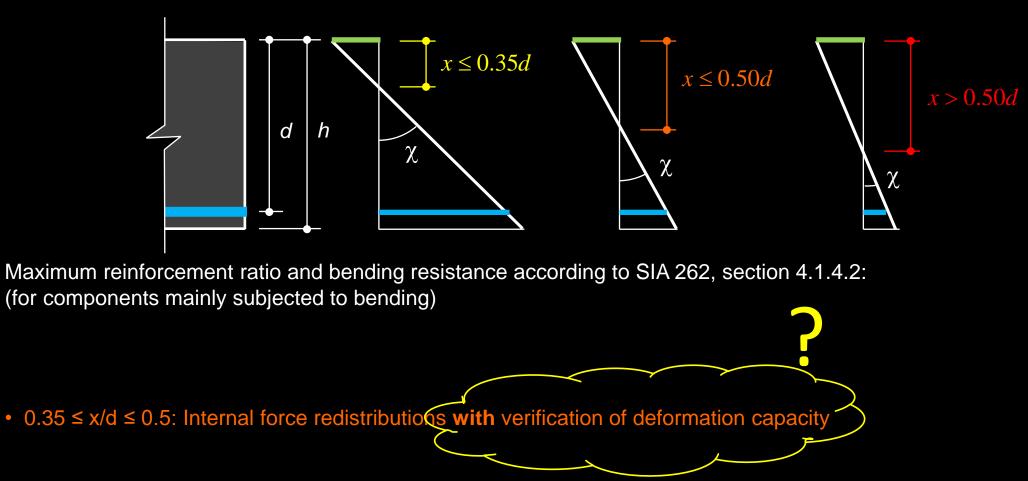
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Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

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Limitation of the compression zone depth according to SIA 262

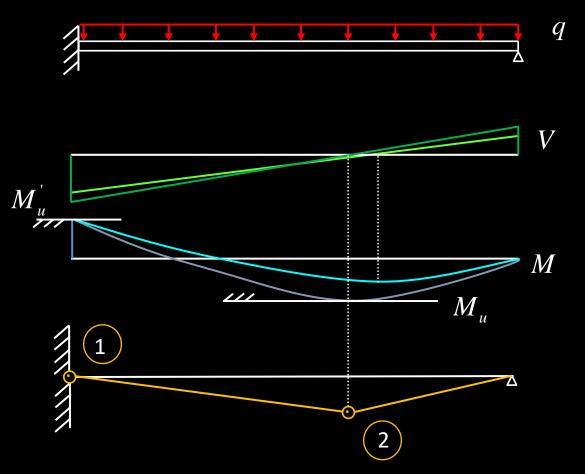
System behaviour (see also [6], page 2.32ff)

Continuous increase of the load *q*:

- \rightarrow Yielding begins first at the fixed-end support, forming a plastic hinge
- \rightarrow The statically indeterminate system turns (for additional loading) into a simple beam

Further load increase is possible until a second plastic hinge is formed in the member (= mechanism):

- \rightarrow Plastic rotation required at the fixed support
- → Rotation demand depending on static system and load configuration
- → Rotation capacity limited by steel elongation and / or concrete compression



Verification = Comparison: Deformation capacity $\Theta_{pu} \leftrightarrow$ Deformation demand $\Theta_{pu,dem}$

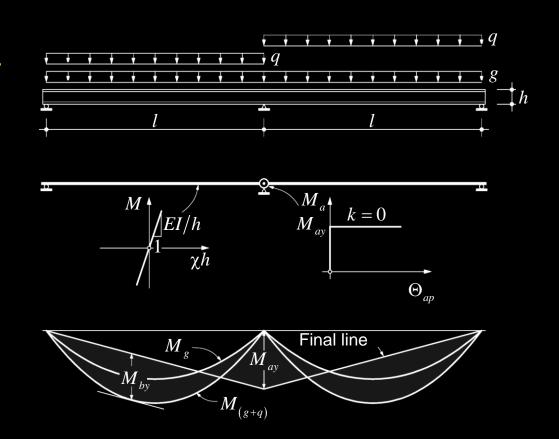
Rotation demand $\Theta_{pu,dem}$ (approximation, example two-span beam) In general, deformation capacity and deformation demand are coupled. The interaction can only be neglected for moderate redistributions. Additional simplifications:

- Constant bending stiffness
- $M-\Theta$ rigid-ideal plastic (no hardening in the plastic hinge)

Therefore, the rotation demand $\Theta_{pu,dem}$ of the intermediate support corresponds to the relative rotation of the two beams over the intermediate support, which can be considered as simply supported beams after reaching M_{ay}

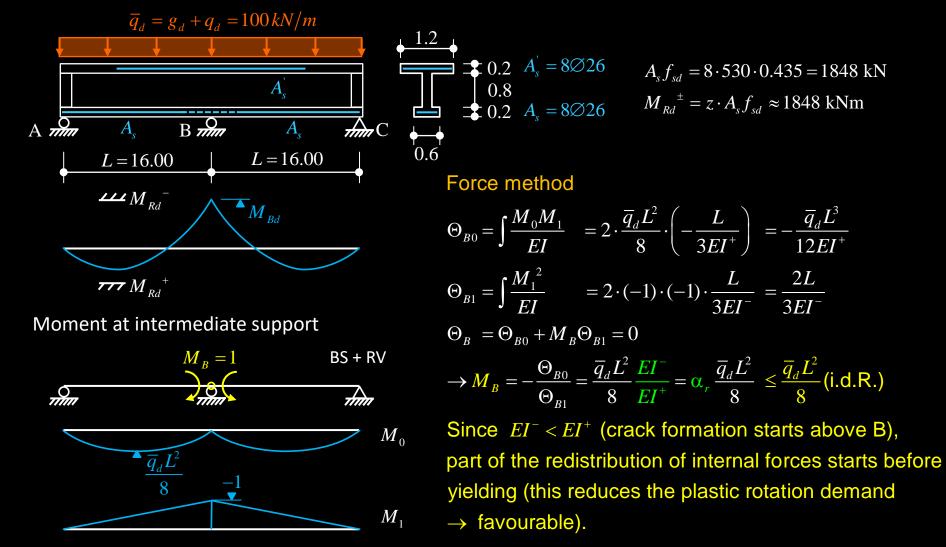
 $(at q = q_y)$:

$$\Theta_{pu,dem} = \frac{\left(q - q_y\right)l^3}{12EI}$$

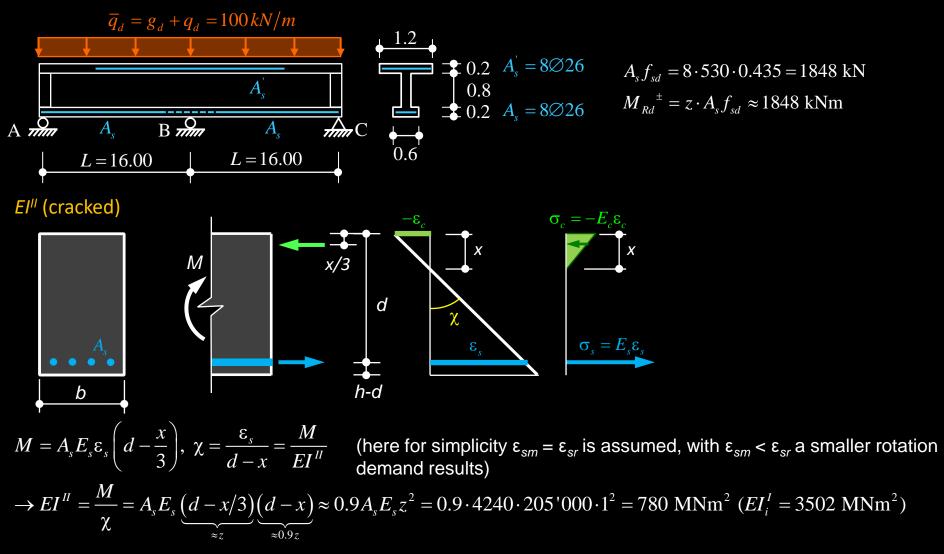


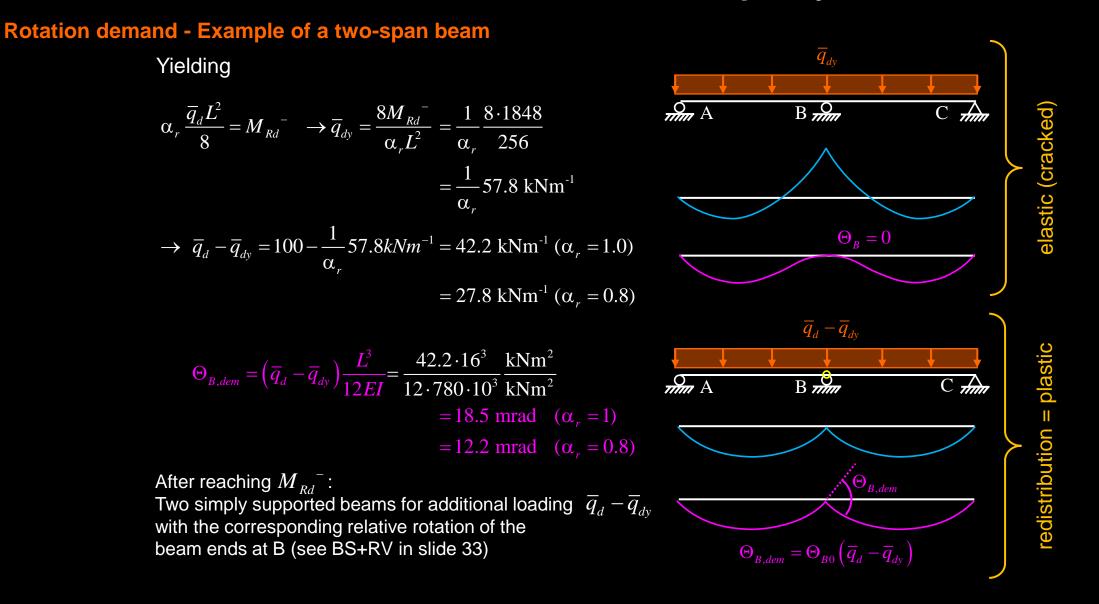
(Two-span beam, first plastic hinge at intermediate support, deformation demand for full load)

Rotation demand - Example of a two-span beam



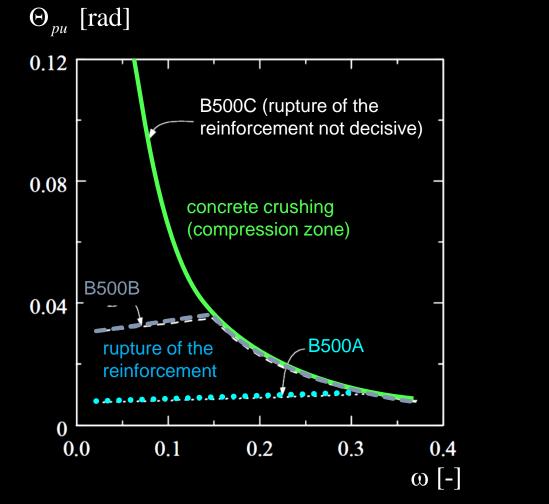
Rotation demand - Example of a two-span beam

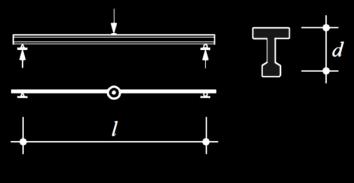




Rotation capacity Θ_{pu} – Basics

Example: Plastic hinge angle as a function of ω (ductility classes A-C, 1999)





Basis of the calculations:

| $f_v = 500 \text{ MPa}$ | l/d = 20 | |
|-------------------------------|--------------------|----|
| $\dot{E}_s = 200 \text{ GPa}$ | $\theta = 45$ | 0 |
| $f_c = 30 \text{ MPa}$ | $\varnothing = 20$ | mm |
| $\varepsilon_{cu} = 5 \%$ | $s_{rm} = 150$ | mm |

Rotation capacity Θ_{pu} (simplified) (see also [6], page 2.32ff)

Limitation of the plastic rotation by the reinforcing steel (rupture of the reinforcement):

 $\Theta_{pus} = L_{pl} \cdot \left(\frac{\varepsilon_{smu}}{d-x} - \frac{\varepsilon_{smy}}{d-x} \right)$

Curvature at onset of yielding Curvature at rupture of the reinforcement

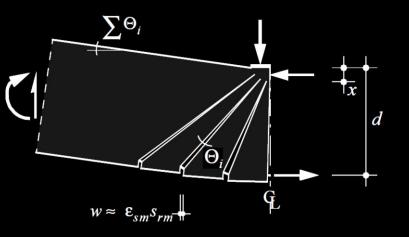
Limitation of the plastic rotation by the concrete (compressive failure):

 $\Theta_{puc} = L_{pl} \left(\frac{\varepsilon_{c2d}}{x} + \frac{\varepsilon_{smy}}{d-x} \right)$

Curvature at onset of yielding Curvature at concrete crushing Rotation per crack: $\Theta_i \approx \frac{\varepsilon_{sm} s_{rm}}{d-x}$

Plastic hinge rotation = sum of the plastic rotations of all cracks from the onset of yielding

el



 L_{pl} Plastic hinge length, depending on load configuration and geometry: region in which the chord reinforcement yields (\rightarrow determine the chord force distribution from the stress field).

 ε_{sr} =

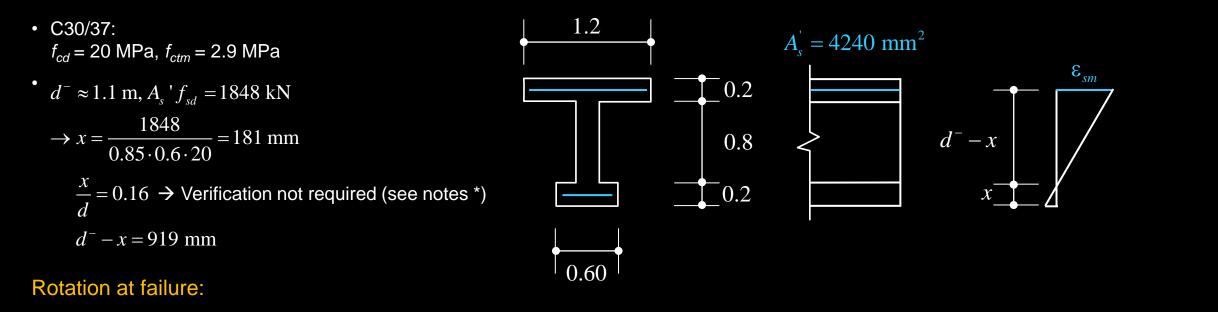
 ϵ_{sr}

- ϵ_{smu} Mean steel elongation when reaching
- ϵ_{smy} Mean steel elongation when reaching

$$\begin{aligned} \varepsilon_{ud} & \sigma_{sr} = f_t & \varepsilon_{sr} \leftrightarrow \varepsilon_{sm} \\ \frac{f_s}{E_s} & \sigma_{sr} = f_s & \text{tension chord mod} \\ \text{(Stahlbeton I)} \end{aligned}$$

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Rotation demand ↔ Rotation capacity (simplified) - Example of a two-span beam



$$\Theta_{puc} = L_{pl} \cdot \left(\frac{\varepsilon_{cu}}{x} - \frac{\varepsilon_{smy}}{d^{-} - x} \right)$$

$$\Theta_{pus} = L_{pl} \cdot \left(\frac{\varepsilon_{smu}}{d^{-} - x} - \frac{\varepsilon_{smy}}{d^{-} - x} \right)$$
with $\frac{\varepsilon_{smy}}{d^{-} - x} =$ Curvature at onset of yielding $= \frac{f_s / E_s - \Delta \varepsilon_0}{d^{-} - x} = 2.3 \text{ mrad/m}, L_{pl} = \text{length plastic hinge} \approx 2d^{-}$

Rotation demand ↔ Rotation capacity (simplified) - Example of a two-span beam

Rotation at failure:

Concrete crushing

$$\Theta_{puc} = L_{pl} \cdot \left(\frac{\varepsilon_{cu}}{x} - \frac{\varepsilon_{smy}}{d^{-} - x}\right) \approx 2 \cdot 1.10 \cdot \left(\frac{0.003}{0.181} - 0.0023\right) = 14.3 \quad \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 31.4 \text{ mrad}$$
$$\rightarrow \Theta_{puc} > \Theta_{B,dem} \rightarrow \text{OK}$$

Steel rupture

rough assumption: $\varepsilon_{smu} \approx 0.5\varepsilon_{ud} = \begin{cases} 22.5\% \text{ (B500B)} \\ 32.5\% \text{ (B500C)} \end{cases}$

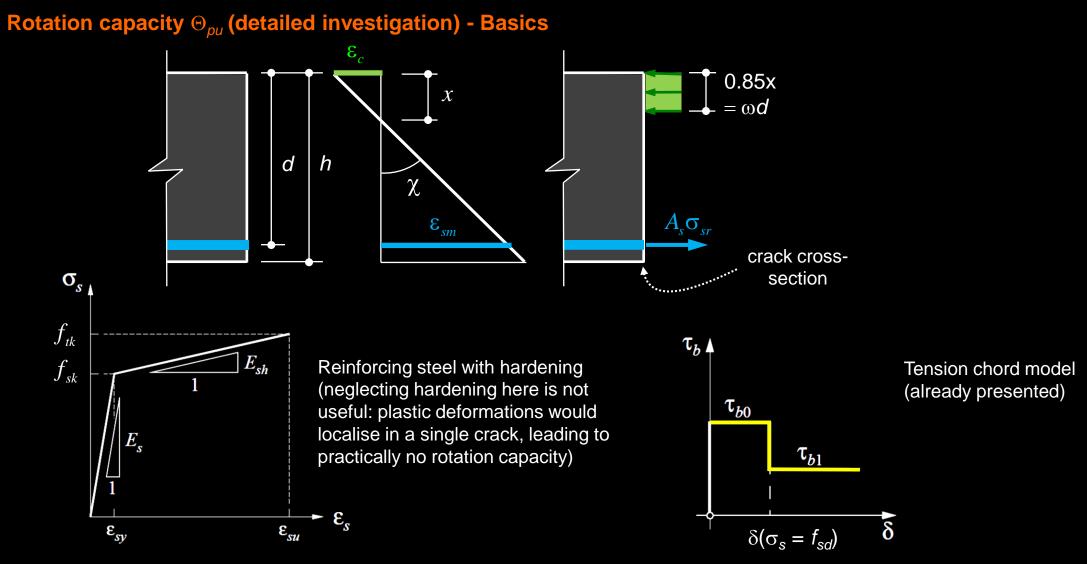
(estimated reduction of elongation at failure due to tension stiffening - see next slides)

$$\Theta_{pus} = L_{pl} \cdot \left(\frac{\varepsilon_{smu}}{d^{-} - x} - \frac{\varepsilon_{smy}}{d^{-} - x}\right) = \begin{cases} 2 \cdot 1.10 \cdot \left(\frac{0.0225}{0.919} - 0.0023\right) = 22.2 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 48.8 \text{ mrad} \text{ (B500B)} \\ 2 \cdot 1.10 \cdot \left(\frac{0.0325}{0.919} - 0.0023\right) = 33.1 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 72.7 \text{ mrad} \text{ (B500C)} \end{cases}$$

$$\rightarrow \Theta_{pus} \ge \Theta_{B,dow} \rightarrow \text{OK}$$

The rotation capacity would be verified.

But: Are the assumptions of L_{pl} , ε_{smu} all right?



Rotation capacity (detailed investigation) - Example of a two-span beam

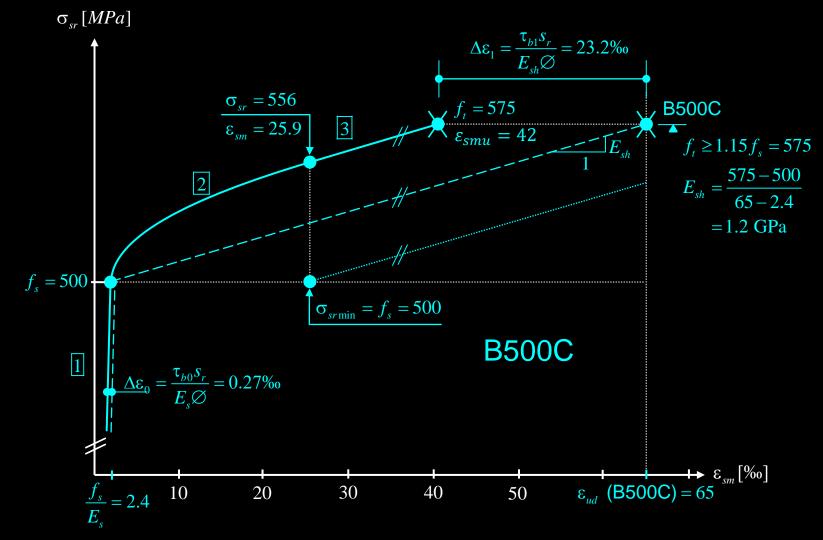
• C30/37: 1.2 $A_{s} = 4240 \text{ mm}^{2}$ $f_{cd} = 20 \text{ MPa}, f_{ctm} = 2.9 \text{ MPa}$ E_{sm} • $d^- \approx 1.1 \text{ m}, A_s' f_{sd} = 1848 \text{ kN}$ 0.2 $\rightarrow x = \frac{1848}{0.85 \cdot 0.6 \cdot 20} = 181 \text{ mm}$ d'-x0.8 $d^{-} - x = 919 \text{ mm}$ 0.2 Equivalent reinforcement ratio (considering x at failure, see notes *): 0.60 $\frac{M_r(d^--x)E_s}{1-n}$

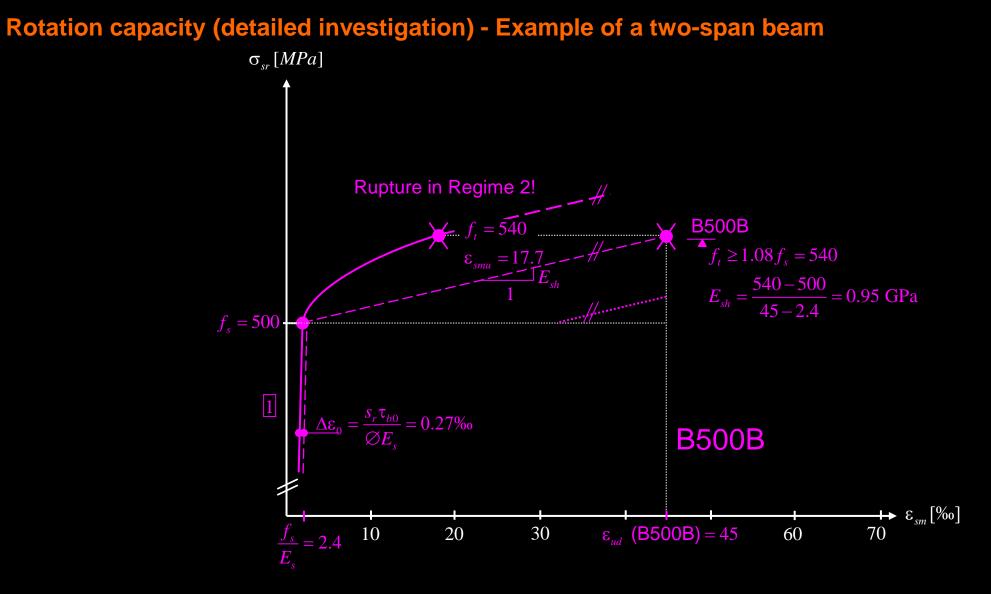
 $s_{rm0} \approx \frac{\emptyset}{4} \cdot \left(\frac{1}{\rho_t} - 1\right) = 292 \text{ mm} \left(\lambda = \frac{1}{2} \dots 1\right)$

 $\rightarrow s_{rm} \approx 250 \text{ mm}$ (spacing of stirrups)

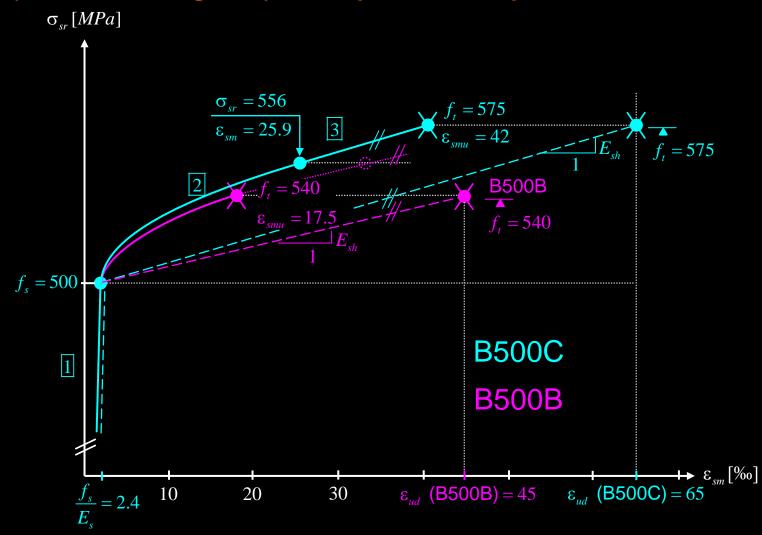
Rotation capacity (detailed investigation) - Example of a two-span beam **Tension chord model** 1 $\sigma_{sr} < f_s$ "elastic" $\emptyset = 26 \text{ mm}$ $\rightarrow \varepsilon_{sm} = \frac{\sigma_{sr}}{E_s} - \frac{\tau_{b0}s_r}{E_s\varnothing} = \frac{\sigma_{sr}}{E_s} - 0.27\%$ $s_{rm} = 250 \text{ mm}$ (spacing of stirrups) $E_{*} = 205 \text{ GPa}$ nackter Stahl $\Delta \varepsilon_0 = \Delta \varepsilon^{1}$ $f_{atm} = 2.9 \text{ MPa}$ 2 $\sigma_{sr} > f_s, \sigma_{smin} < f_s \rightarrow \dots$; Transition to regime 3 at $\tau_{h0} = 2 f_{ctm} = 5.8 \text{ MPa}$ "partially yielded" $\sigma_{smin} = \sigma_{sr} - \frac{2\tau_{b1}s_r}{\varnothing} = \sigma_{sr} - 56 \text{ MPa} \stackrel{!}{=} f_s$ $\tau_{b1} = \overline{1f_{ctm}} = 2.9$ MPa $\rightarrow \sigma_{sr} = f_s + 56 \text{ MPa} \rightarrow B500B \text{ stays at regime } 2$ $3 \sigma_{smin} > f_s \rightarrow \varepsilon_{sm} = \frac{f_s}{\underbrace{E_s}} + \frac{\sigma_{sr} - f_s}{\underbrace{E_{sh}}} - \frac{\tau_{b1}s_r}{\underbrace{E_{sh}}}$ "fully yielded" **B500C**: $\varepsilon_{sm} (\sigma_{sr} = f_s) = 2.43 - 0.27 = 2.16\%$ $\varepsilon_{smin} (\sigma_{smin} = f_s) = 25.9\%$ (3) with $\sigma_{sr} = 556$ MPa) 23‰ (B5000 $\epsilon_{sm} (\sigma_{sr} = f_t) = 65 - 23 = 42\% = \epsilon_{sm}$ **B500B**: $\varepsilon_{sm}(\sigma_{sr} = f_s) = 2.16\%$ $\sigma_{sr} = f_{sd} + E_{sh}$ $\varepsilon_{smu} = 17.7\%$ (Regime 2) with $\sigma_{sr} = f_t$, does not reach regime 3)

Rotation capacity (detailed investigation) - Example of a two-span beam

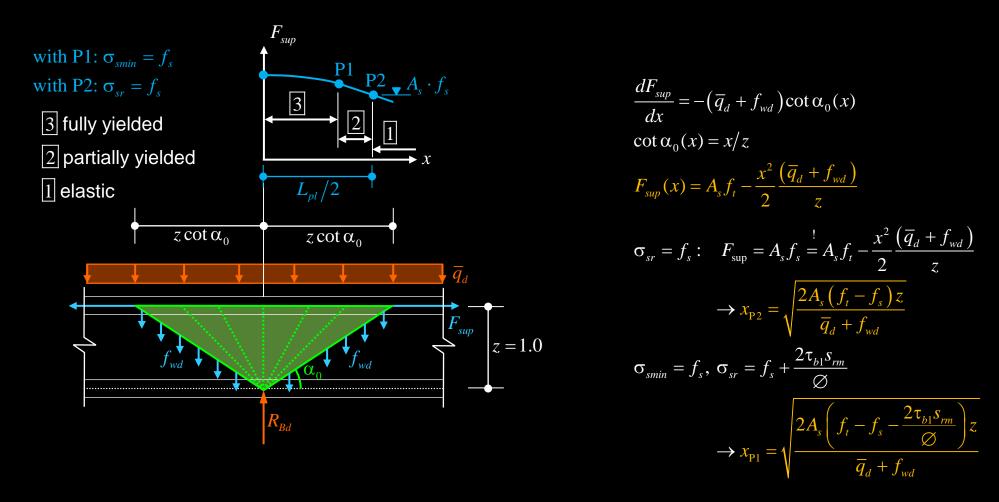




Rotation capacity (detailed investigation) - Example of a two-span beam



Rotation capacity (detailed investigation) - Example of a two-span beam Plastic hinge length \rightarrow Distribution of the top chord force F_{sup} determined from a stress field

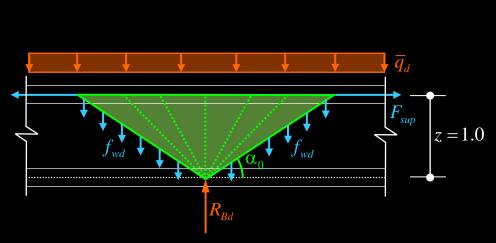


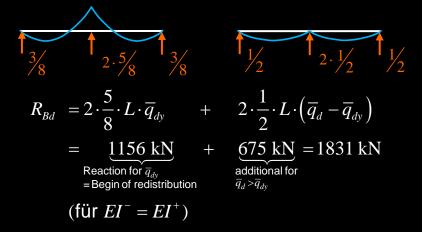
Rotation capacity (detailed investigation) - Example of a two-span beam Plastic hinge length \rightarrow Distribution of the top chord force F_{sup} determined from a stress field

 R_{Bd} increases during the redistribution, x_P thus decreases (large gradient of *M* is unfavourable for the rotational capacity, since a stronger localization of deformations occurs):

 R_{Bd} (and thus x_P) also depends on the choice of the compression field inclination α_0 :

 $R_{Bd} = 2z \cot(\alpha_0)(\overline{q}_d + f_{wd}) \rightarrow (\overline{q}_d + f_{wd}) = \frac{R_{Bd}}{2z \cot(\alpha_0)}$ large $\alpha_0 \rightarrow x_{P1}, x_{P2}$ small, small $\alpha_0 \rightarrow x_{P1}, x_{P2}$ large





- \rightarrow Several assumptions are necessary to determine the deformation capacity
- → Rough estimation, not exact calculation!

Rotation capacity (detailed investigation) - Example of a two-span beam

Plastic hinge length \rightarrow Distribution of the top chord force F_{sup} determined from a stress field

$$\Rightarrow \text{Assumption:} \qquad R_{Bd} \approx 1500 \text{ kN, } \cot(\alpha_0) = 1.5 \ (\alpha_0 = 33.5^\circ), \ \left(\overline{q}_d + f_{wd}\right) = \frac{1500}{2\cot(\alpha_0)} = \frac{1500}{3} = 500 \text{ kNm}^{-1}$$

B500B:
$$x_{p_2} = \sqrt{\frac{2 \cdot 4240(540 - 500) \cdot 1000}{500}} = 823 \text{ mm} ("L_{p_l}/2")$$

B500C: $x_{p_2} = \sqrt{\frac{2 \cdot 4240(575 - 500) \cdot 1000}{500}} = 1127 \text{ mm} ("L_{p_l}/2")$
 $x_{p_1} = \sqrt{\frac{2 \cdot 4240(575 - 556) \cdot 1000}{500}} = 571 \text{ mm}$
 $\varepsilon_{smu} [\%_{o}] = \sqrt{\frac{41}{500}} = \frac{10000}{500} = 571 \text{ mm}$
 $\varepsilon_{smu} [\%_{o}] = \frac{41}{100} = \frac{10000}{500} = 571 \text{ mm}$
 $\varepsilon_{smu} [\%_{o}] = \frac{41}{100} = \frac{10000}{500} = 571 \text{ mm}$
 $\varepsilon_{smu} [\%_{o}] = 10.5\% \text{ (averaged over } L_{pl} = 1.65 \text{ m})$
B500C: $\overline{e}_{smu} \approx \frac{2}{L_{pl}} \cdot \left(\int_{0}^{x_{p1}} \varepsilon_{sm}(x) \cdot dx + \int_{x_{p1}}^{x_{p2}} \varepsilon_{sm}(x) \cdot dx\right)$
 $= 2.16 = \frac{100000}{500} = \frac{10000}{500} = 10.5\% \text{ (averaged over } L_{pl} = 2.25 \text{ m}$

 $L_{Pl}/2$

Rotation demand and rotation capacity (detailed investigation) - Example two-span beam

Plastic rotation at failure

Concrete crushing

$$\Theta_{puc} = L_{pl} \cdot \left(\frac{\varepsilon_{cu}}{x} - \frac{\varepsilon_{smy}}{d^{-} - x}\right) \approx 2 \cdot 1.10 \cdot \left(\frac{0.003}{0.181} - 0.0023\right) = 14.3 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 31.4 \text{ mrad}$$
$$\rightarrow \Theta_{puc} > \Theta_{B,req} \rightarrow \text{OK}$$

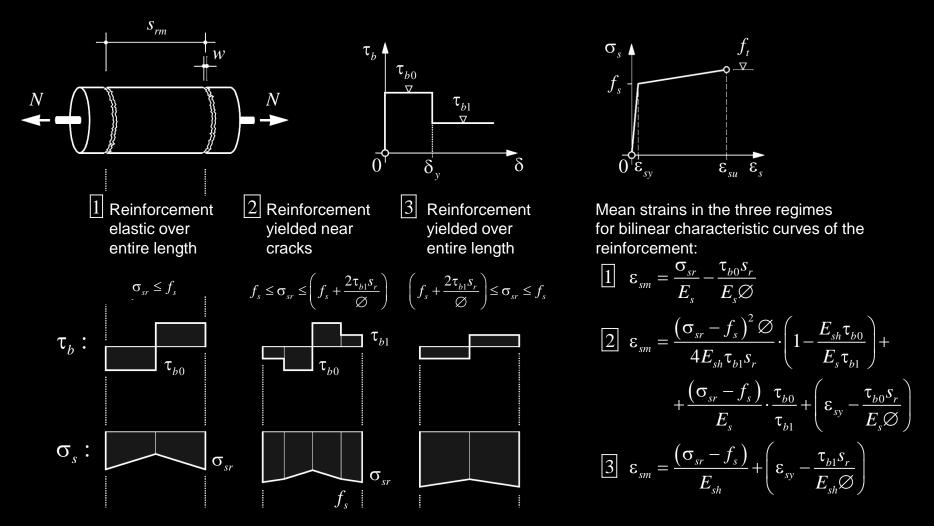
Steel rupture

Rough assumption:
$$\varepsilon_{smu} \approx 0.5\varepsilon_{ud} = \begin{cases} 22.5\% \text{ with } L_{pl} = 2.2 \text{ m} (B500B) \\ 32.5\% \text{ with } L_{pl} = 2.2 \text{ m} (B500C) \end{cases}$$
More detailed investigation: $\overline{\varepsilon}_{smu} = \begin{cases} 10.5\% \text{ with } L_{pl} = 1.65 \text{ m} (B500B) \\ 24.1\% \text{ with } L_{pl} = 2.25 \text{ m} (B500C) \end{cases}$ $\overline{\varepsilon}_{smu} \approx 0.37 \cdot \varepsilon_{ud}, \ L_{pl} \approx 2.0 \cdot d \end{cases}$

$$\Theta_{pus} = L_{pl} \cdot \left(\frac{\overline{\varepsilon}_{smu}}{d^{-} - x} - \frac{\varepsilon_{smy}}{d^{-} - x}\right) = \begin{cases} 1.65 \cdot \left(\frac{0.0105}{0.919} - 0.0023\right) = 15.1 \text{ mrad } (B500B) < \Theta_{B,req} = 18.5 \text{ mrad } (\alpha_{r} = 1) \\ \text{not fulfilled!} \end{cases}$$

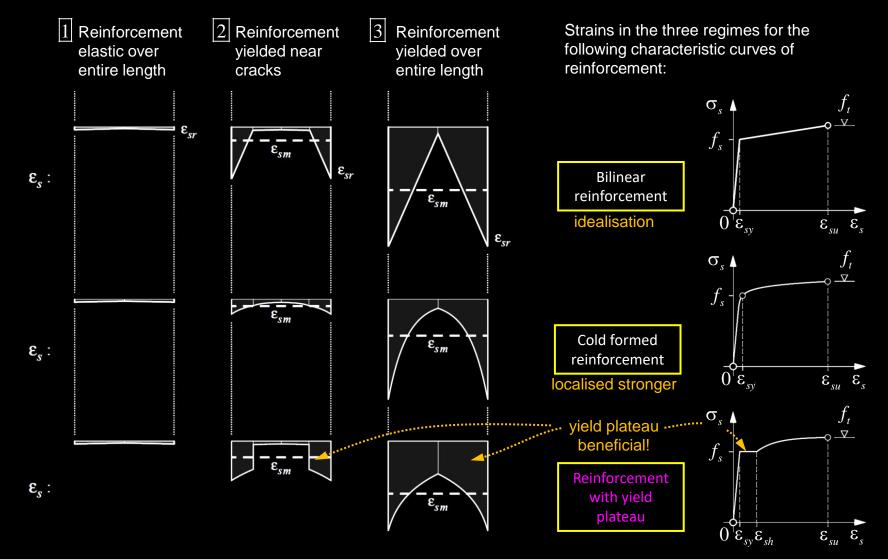
$$2.25 \cdot \left(\frac{0.0241}{0.919} - 0.0023\right) = 53.8 \text{ mrad } (B500C) \gg \Theta_{B,req} = 18.5 \text{ mrad } (\alpha_{r} = 1) \\ \text{ok (no problem)} \end{cases}$$

Additional considerations: ratio of mean strain to maximum strain in the cracks considering bond



[Alvarez 1999]

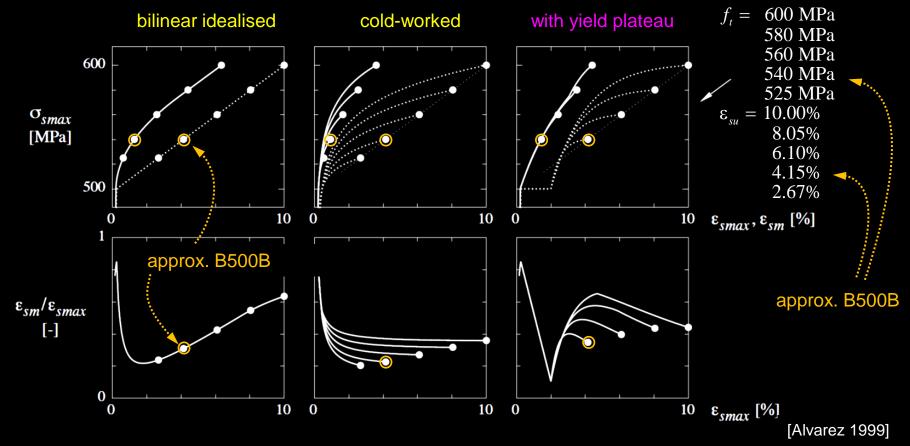
Additional considerations: influence of the reinforcement hardening properties



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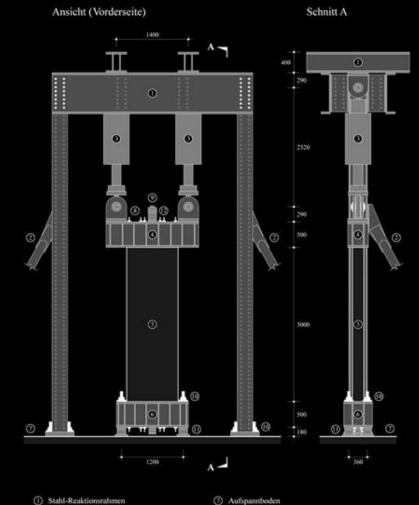
Additional considerations: influence of the reinforcement hardening properties

- → Reinforcement with a yield plateau is more favourable than cold-formed reinforcement, especially in case of failure in regime 2 (yield plateau contributes as an "additional" strain over the entire yielded area)
- → The bilinear idealization overestimates the deformation capacity for a reinforcement with high ductility



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Tension experiments – Dr. M. Alvarez: Test setup



(8) Verankerungen der Anschlussstäbe

(1) Kraftmessdosen (1 MN Nennbereich)

③ Kraftmessdose (2 MN Nennbereich)

(1) Vorspannstangen (Verankerungen im Aufspannboden)

③ Verankerung des Vorspannkabels

- Stahl-Reaktionsrahmen
 Druckstreben (Knicksicherung)
- ③ Servohydraulische Pressen (2 MN Kapazität)
- ④ Oberer Krafteinleitungsträger
- ③ Versuchskörper
- (6) Unterer Krafteinleitungsträger

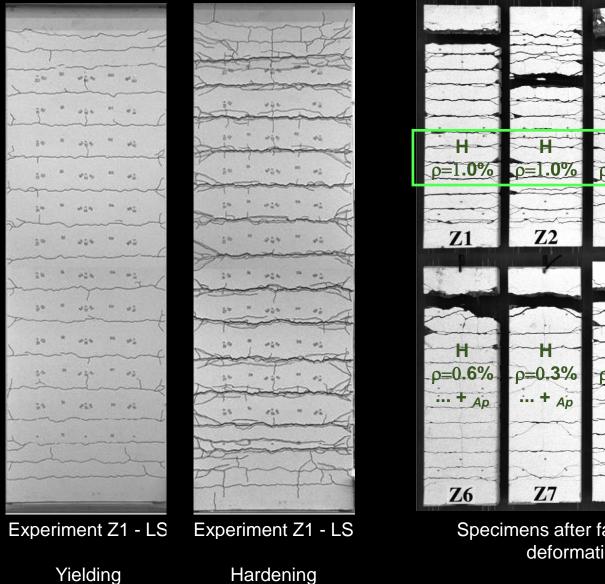
3000 220 A_s A_p

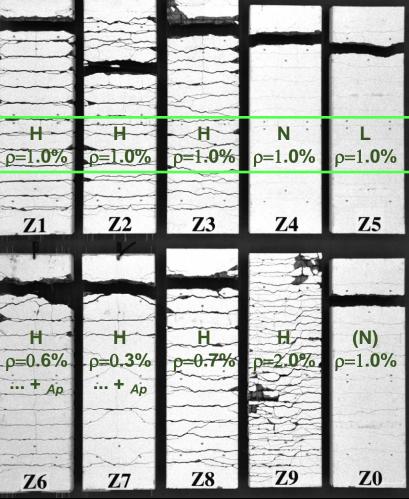
Bild 1.1 – Konzept der Zugversuche, [mm].

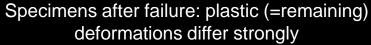
| Versuchskörper | Z1 | Z2 | Z3 | Z4 | Z5 | Z6 | Z 7 | Z8 | Z9 | |
|--|-------------|----|-----------|----|----|---------------|------------|------|------|--|
| Längsbewehrung A_s [mm ²] | 2156 | | | | | 1232 | 616 | 1540 | 4312 | |
| Vorspannbewehrung A_p [mm ²] | 0 | | | | | 1050 0 | | | | |
| Betonstahlqualität | Н | | | Ν | L | | Н | | | |
| Bügelbewehrung [mm] | Ø 8 @ 200 0 | | Ø 8 @ 200 | | | | | | | |
| Würfeldruckfestigkeitdes Betons f_{cw} [MPa] | 50 | 90 | | 50 | | | | | | |

| H: ε _{su} = 14.6% | $f_t / f_s = 1.26$ |
|-----------------------------------|----------------------|
| Ν: ε _{su} = 3.8% | $f_{t}/f_{s} = 1.05$ |
| L: ε _{su} = 3.1% | $f_t / f_s = 1.06$ |

Tension experiments – Dr. M. Alvarez: Crack patterns at failure





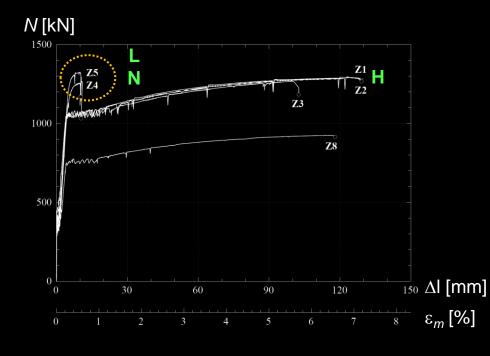


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Tension experiments – Dr. M. Alvarez: Test results

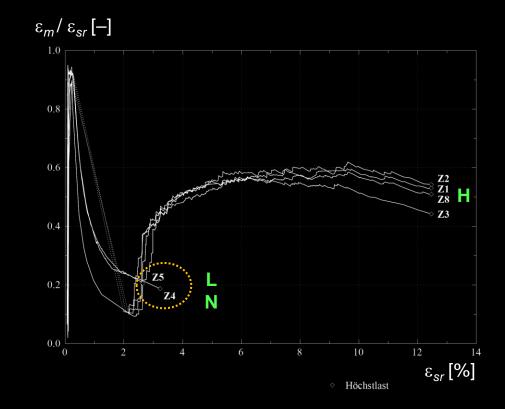
Load-deformation behaviour considering bond

→ Deformation capacity severely impaired for reinforcement with low ductility (failure deformation and hardening!)



Ratio of average elongation to maximum elongation at the cracks considering bond

→ Good agreement with tension chord model (almost identical if the real bare steel curve is taken into account)



Deformation capacity

Summary

- The concrete strength should be reduced in plastic analysis depending on the cracking state of the structure and on the material brittleness.
- The concrete contribution in tension between two cracks stiffens the response of bonded reinforcement with respect to bare (unbonded) reinforcement. This tension-stiffening effect affects the serviceability response of the structure but also reduces the deformation capacity of the reinforcement. Assuming simplified bond relationships (as e.g. in the Tension Chord Model) is sufficient for modelling tension-stiffening.
- Deformation capacity and deformation demand are coupled. The interaction can only be neglected for moderate redistributions of the internal forces.
- The *deformation demand* can be determined approximately with reasonable effort using simplified assumptions (constant bending stiffness of the elastic areas, rigid-ideal plastic *M*- Θ relationships of the plastic hinges).
- Even with complex calculations, the *deformation capacity* can only be roughly estimated because it depends on several effects and assumptions that cannot be precisely quantified:
 - Bond behaviour, in particular, crack spacing
 - Mechanical properties of the reinforcement (hardening ratio and deformation of failure, with or without yield plateau)
 - Force flow in the area of plastic hinges, in particular, variation of the force in the tension chord
 (→ the mean deformations averaged over the length of the plastic hinge are smaller than the mean deformation of a tension chord under constant tensile force!)
- In practice, it is therefore advisable to avoid the verification of the deformation capacity for new structures whenever possible (complying with the condition x/d < 0.35). Otherwise, it is often easier to ignore the redistribution of internal forces, i.e. to verify the structural safety for the elastic stresses including restraint stresses (even if the estimation of the restraints is also time-consuming and requires assumptions).
- If the deformation capacity needs to be verified (e.g. for existing structures), engineering judgement must be applied. The decisive parameters should be accounted for as accurately as possible (reinforcement: determine hardening characteristics, not just *f*_s).