2 In plane loading – walls and beams

2.3 Compatibility and deformation capacity

Learning objectives

Within this chapter, the students are able to:

- **EXP** determine the behaviour of concrete as a function of the compressive strength and the cracking state.
- **•** recognise the assumption of limit analysis methods for the materials having sufficient deformation capacity to reach the plastic solution without rupturing, and the existence of approaches to verify the deformation capacity of the materials.
- **•** evaluate plastic redistributions of internal forces in hyperstatic systems (beams and frames) subjected either to external loads or imposed deformations, and calculate the deformation demand in elements subjected to bending or normal actions.
- **EXEDEE FIELD EXE** estimate the deformation capacity of a structure subjected to bending or normal actions.
	- \circ explain the tension-stiffening effect and how it affects the structural behaviour.
	- \circ illustrate the main assumptions and behaviour of bonded reinforcement according to the Tension Chord Model.

2 In plane loading – walls and beams

2.3 Compatibility and deformation capacity

A) Behaviour of concrete in compression

Main factors influencing the equivalent strength to be considered in plastic calculations

Strain softening after peak strength (material effect)

The concrete brittleness (i.e. relative amount of softening) increases with the compressive strength and also the reduction of the strength to be accounted for (η_{fc}) .

Influence of transverse cracking on concrete strength (structural effect)

Reduction factor to account for this effect (*k^c*) can be determined in a more refined manner based on the state of deformations (see following slides).

Dependence of the concrete compressive strength and shear resistance on the strain state

Tests have shown that the compressive strength in membrane elements is reduced by (imposed) transverse strains.

In 1986, Vecchio and Collins proposed reducing the compressive strength by a factor $1/(0.8\!+\!170\cdot\!\epsilon_{_{1}})\!\leq\!1$. The setting \Box (assuming "average" concrete stresses). This also takes implicitly other effects into account.

In 1998, Kaufmann proposed to consider additionally the (already known) inversely proportional increase of the compressive strength with the cylinder compressive strength:

2/3 $f_{c,cyl}^{2/5}$ $f_{ce} = \frac{\partial c_{c,eyl}}{\partial .4 + 30 \cdot \varepsilon_1}$ $+30. \varepsilon$

On the basis of this and other work, SIA 262 has introduced the following coefficient for the verification of webs of beams:

1

 $\frac{1}{1}$ and $\frac{1}{2}$ are This can be applied in a more general $k_c = \frac{1}{1,2+55\epsilon_1} \leq 0.65$ way to any structural member when removing the 0.65 upper-bound

 $3 \cup 4$ $\alpha \cup 1$ 8 Assumption $\varepsilon_3 = -0.002$ 1 X Q α α $\epsilon_{\mathbf{x}} - \epsilon_{\mathbf{3}}$ $(\epsilon_{\mathbf{x}} - \epsilon_{\mathbf{3}})\cot^2 \alpha$ α $(\varepsilon_{x} - \varepsilon_{3}) \cot \alpha$ $\gamma/2$
 x Assumption $\varepsilon_3 = -0.002$
 x Assumption $\varepsilon_3 = -0.002$
 3 $\left(\varepsilon_x - \varepsilon_3\right) \cot^2 \alpha$
 $\left(\varepsilon_x - \varepsilon_3\right) \cot^2 \alpha$ 2 1 \mathbf{v}_x \mathbf{v}_x \mathbf{v}_x 1 1 $k_c(\varepsilon_x, \alpha) = \frac{1}{1.2 + 55 \cdot \varepsilon_x} \le 0.65$ $\varepsilon_1 = \varepsilon_x + (\varepsilon_x + 0.002) \cdot \cot^2 \alpha$ $\mathbf{z}_1 = \mathbf{\varepsilon}_x + (\mathbf{\varepsilon}_x + 0.002) \cdot \cot^2 \alpha$ **x x** $\frac{1}{25 \cdot \varepsilon_1} \le 0.65$ $\varepsilon_1 = \varepsilon_x + (\varepsilon_x + 0.002) \cdot \cot^2 \alpha$
 $\frac{1}{\alpha - 15^{\circ}}$ $\frac{20^{\circ}}{30^{\circ}}$ $\frac{35^{\circ}}{35^{\circ}}$ $\frac{45^{\circ}}{3}$ $\frac{\varepsilon_x - \varepsilon_3}{3}$ $\begin{array}{c|c|c|c} \multicolumn{3}{c|}{} & \multic$ [−]10 E_1 [%o] ϵ \mathbf{S} $\gamma/2$ \blacktriangle \times X $3 \cup 1$ $\alpha \cup 1$ ϵ Q -2 0 2 4 6 8 10 ε_{x} [‰]

Concrete compressive strength and shear resistance as a function of the strain state

Concrete compressive strength and shear resistance as a function of the strain state

 \to k_c f_c is largely reduced for flat inclinations of the compression field and for plastic strains of the tension chord

Concrete compressive strength and shear resistance as a function of the strain state

- \to k_c f_c is largely reduced for flat inclinations of the compression field and for plastic strains of the tension chord
- \rightarrow Concrete compressive stresses increase sharply with flat inclinations (see above)
- \rightarrow Very flat inclinations do not make sense when dimensioning, but are often necessary when assessing old bridges
-

2 In plane loading – walls and beams

2.3 Compatibility and deformation capacity

B) Behaviour of bonded reinforcement

 \rightarrow ODE of 2nd order

Differential equations of bond

General bond-slip law

Equilibrium of an element with length *dx*:

$$
\frac{d\sigma_c}{dx} = -\frac{\cancel{\mathcal{D}}\pi\tau_b + q_x}{A_c(1-\rho)};
$$
\n
$$
\frac{d\sigma_s}{dx} = \frac{4\tau_b}{\cancel{\mathcal{D}}}
$$
\n
$$
\rightarrow
$$
 ODE of 1st order

Considering linear elastic material behaviour:

 $(1-\rho)$ $2c \tA - \alpha$ 2 α Γ Λ Γ $4\tau_{\kappa}$ $\varnothing \pi \tau_{\kappa} + q_{\kappa}$ $\mathcal{O}E$, $A E$ _c $(1-\rho)$ b , \mathcal{P}^{\prime} \mathcal{P}^{\prime} \mathcal{V}^{\prime} q_x *s c c* $d^2\delta$ $4\tau_h$ $\mathcal{Q}\pi\tau_h + q_x$ dx^2 QE A E $(1-\rho)$ δ 4τ, \varnothing πτ, + *a* $=$ $-\frac{v}{r}$ + $-\frac{v}{r}$ + $\frac{v}{r}$ $-\rho$)

Simplified bond-slip law, used in TCM

Equilibrium of an element with length *dx*:

 \rightarrow linear (integrate once) $\int_{c} (1-\rho)$ *dx*

Considering linear elastic material behaviour:

Behavior of the bonded reinforcement – Ten
\n**ifferential equations of bond**
\neneral bond-slip law #
\nEquilibrium of an element with length dx:
\n
$$
\frac{d\sigma_x}{dx} = -\frac{\varphi \pi \tau_b + q_x}{A_c (1-\rho)}; \qquad \frac{d\sigma_x}{dx} = \frac{4\tau_b}{\varphi} \qquad \rightarrow \text{ODE of 1st order}
$$
\nConsidering linear elastic material behaviour:
\n
$$
\frac{d^2\delta}{dx^2} = \frac{4\tau_b}{\varphi E_x} + \frac{\varphi \pi \tau_b + q_x}{A_c E_c (1-\rho)} \qquad \rightarrow \text{ODE of 2nd order}
$$
\n
$$
\text{implified bond-slip law, used in TCM}
$$
\n
$$
\frac{d\sigma_x}{dx} = -\frac{\varphi \pi \tau_b + q_x}{A_c (1-\rho)} = \text{const}^*; \qquad \frac{d\sigma_x}{dx} = \frac{4\tau_b}{\varphi} = \text{const}^* \qquad \rightarrow \text{linear (integrate once)}
$$
\n
$$
\text{Considering linear elastic material behaviour:}
$$
\n
$$
\frac{d^2\delta}{dx^2} = \frac{4\tau_b}{\varphi E_x} + \frac{\varphi \pi \tau_b + q_x}{A_c E_c (1-\rho)} = \text{const}^* \qquad \rightarrow \text{quadratic (integrate twice)}
$$
\n
$$
\frac{d^2\delta}{dx^2} = \frac{4\tau_b}{\varphi E_x} + \frac{\varphi \pi \tau_b + q_x}{A_c E_c (1-\rho)} = \text{const}^* \qquad \rightarrow \text{quadratic (integrate twice)}
$$
\n
$$
\text{equation: } \text{Equation (Charar of Concrete Structures and Bridge Design | Advanced S
$$

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View of a tension chord (total cross-section A_c), reinforced with bar with diameter Ø ([6], page 3.5f)

Concrete stress in the middle of the element with length s_{r0} (maximum crack spacing) is $\sigma_c = f_{ctm}$ i.e. another crack could form there.

$$
S_{rm0} \approx \frac{\varnothing}{4} \cdot \left(\frac{1}{\rho_t} - 1\right)
$$

Thus the minimum crack spacing is:

 $S_{r,min} = S_{r0}/2$

Generally, the crack spacing varies with parameter λ :

$$
s_r = \lambda \cdot s_{r0} \qquad \left(\frac{1}{2} < \lambda < 1\right)
$$

 \rightarrow Theoretical limits of the crack spacing with fully developed crack pattern!

SN: If the cracks form because of applied loads, the fully developed crack pattern forms at once (theoretically).

Increase of normal force after crack formation $N > N_r$ ([6], page 3.5f)

Increase of normal force after crack formation $N > N_r$ ([6], page 3.5f)

Increase of normal force after crack formation $N>N_r$ ([6], page 3.5f)

N-ε- and σ_{sr}-ε-diagrams : Reduction of the elongation of the bare steel by $\Delta \varepsilon$ $(\Delta \varepsilon$ remains constant until yielding).

NB: Good approximation for w_r (small ρ)

$$
\frac{\emptyset/4\rho}{2E_s} \left(\frac{N}{A_s} - \frac{f_{ctm}}{4\rho}\right) \leq w_r \leq \frac{\emptyset/4\rho}{E_s} \left(\frac{N}{A_s} - \frac{f_{ctm}}{2\rho}\right)
$$

Concrete stresses remain constant after cracking. Steel stresses keep increasing.

 $\varepsilon_{cm} = \lambda f_{ctm}/(2E_c)$

Steel elongation at crack Average concrete elongation

$$
\varepsilon_{sr}=\sigma_{sr}/E
$$

Mean steel elongation

$$
\varepsilon_{sm} = \frac{\sigma_{sr}}{E_s} - \frac{\tau_{b0}}{\phi} \frac{s_r}{E_s} = \frac{\sigma_{sr}}{E_s} - \frac{\left(\lambda f_{ctm} (1 - \rho)\right)}{2\rho E_s}
$$

Crack widths: Difference of the mean steel and concrete strains multiplied by $s_r(\lambda = 0.5...1)$:

$$
w_r = s_r \left[\frac{\sigma_{sr}}{E_s} - \frac{\lambda f_{ctm} (1 - \rho)}{2 \rho E_s} - \frac{\lambda f_{ctm}}{2 E_c} \right] = \frac{\lambda s_{r0} (2 \sigma_{sr} - \lambda \sigma_{sr0})}{2 E_s}
$$

with $\sigma_{sr} = N/A_s$

$$
\frac{s_{r0}}{2 E_s} \left(\sigma_{sr} - \frac{\sigma_{sr0}}{4} \right) \leq w_r \leq \frac{s_{r0}}{E_s} \left(\sigma_{sr} - \frac{\sigma_{sr0}}{2} \right)
$$

 $N_{sy} \leq N = N_{sr} \leq N_{su}$

 Elastic reinforcement over entire crack element

$$
\sigma_{sr} \leq f_{sy}
$$

our of the b
ution for a biline
prement over er
ent yields near creacks
 $\frac{2\tau_{b1}s_r}{\varnothing}$ **Behaviour**

sed form solution

Delastic reinforcem

rack element

For $\leq f_{sy}$

Delastic between crack

Reinforcement yie

lastic between crack
 $\frac{1}{s} \leq \sigma_{sr} \leq \left(f_{sy} + \frac{2\tau_{b1} s_r}{\varnothing}\right)$ **Reinforcement yields near**
 satic between cracks
 $\leq \sigma_{sr} \leq \left(f_{sy} + \frac{2\tau_{b1} s_r}{\varnothing}\right)$
 Reinforcement yields over ack element
 $\frac{1}{\varnothing} + \frac{2\tau_{b1} s_r}{\varnothing} \leq \sigma_{sr} \leq f_{su}$
 ETH Reinforcement yields near cracks, elastic between cracks

 $\mathcal{I}_{h1} S_r$ $\sigma_{\sim} \leq |t_{+} + \frac{b_1 + b_2}{2}|$

3 Reinforcement yields over entire crack element

\n- Q Reinforcement yields near cracelastic between cracks
\n- $$
f_{sy} \leq \sigma_{sr} \leq \left(f_{sy} + \frac{2\tau_{b1} s_r}{\varnothing} \right)
$$
\n- Q Reinforcement yields over entityer
\n- Grack element
\n- $$
\left(f_{sy} + \frac{2\tau_{b1} s_r}{\varnothing} \right) \leq \sigma_{sr} \leq f_{su}
$$
\n- 10.2024
\n

Behavior of the bonded reinforcement – Tension chord model (SBI)	
Closed form solution for a bilinear steel stress-strain relationship	\n $\frac{d}{dx} \sum_{m \neq 0} \frac{d}{dx} \sum_{m \neq 0} \frac{$

$$
\sigma_{sr} = f_{sy} + E_{sh} \left(\varepsilon_{sm} - \frac{f_{sy}}{E_s} \right) + \frac{\tau_{b1} s_r}{\varnothing}
$$
\nwhere $\text{steel} - \Delta \varepsilon_1$ "
\n
$$
\varepsilon_{sm} = \varepsilon_{sy} + \frac{\left(\sigma_{sr} - f_{sy} \right)}{E_{sb}} - \frac{\tau_{b1} s_r}{E_{sb} \varnothing}
$$
\n
$$
\Delta \varepsilon_1 = \frac{\tau_{b1} s_r}{E_{sh} \varnothing}
$$

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Constitutive relationship of the bonded reinforcement (tension chord model with bilinear bare reinforcement):

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Load-deformation behaviour considering bond (influence at high loads)

- \rightarrow No influence on tensile resistance
- \rightarrow Stiffer behaviour than bare steel

Ratio of average elongation to maximum elongation at the cracks considering bond

- \rightarrow Heavy drop after onset of yielding
- \rightarrow Pronounced influence on ductility!

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Behaviour of the bonded reinforcement

Application to loading cases different than uniaxial tension

Simple bending (SB I): Elastic bending stiffness – tensile stiffness [6], page 2.16f

Behaviour of the bonded reinforcement

Tension stiffening for non-stabilised crack patterns (pull-out model)

for $p \leq p_{cr} \approx 0.6\%$ \rightarrow Reinforcement is NOT able to carry the cracking load without yielding and the tension chord model is not applicable. Cracks are controlled by other reinforcement and a stabilized crack pattern is not generated.

- **-** A pull-out tension stiffening model can be formulated for these situations by assuming (a) independent cracks and (b) the same bond slip model as for the tension chord model.
- A certain crack spacing (l_{av}) should be assumed to compute the average reinforcement strain.

2 In plane loading – walls and beams

2.3 Compatibility and deformation capacity

C) Deformation capacity of beams

Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

• x/d ≤ 0.35: Internal force redistributions **without** verification of deformation capacity $x / d \leq 0.35 \rightarrow \omega \leq 0.298 \rightarrow M_{Rd} \leq bd^2 f_{cd} \omega \cdot (1 - \omega/2) = 0.253 \cdot bd^2 f_{cd}$

Limitation of the compression zone depth according to SIA 262

Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

• 0.35 ≤ x/d ≤ 0.5: Internal force redistributions **with** verification of deformation capacity $x / d \leq 0.50 \rightarrow \omega \leq 0.425 \rightarrow M_{Rd} \leq bd^2 f_{cd} \omega \cdot (1 - \omega/2) = 0.335 \cdot bd^2 f_{cd}$

Limitation of the compression zone depth according to SIA 262

Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

• **x/d > 0.50: is to be avoided**

Limitation of the compression zone depth according to SIA 262

Maximum reinforcement ratio and bending resistance according to SIA 262, section 4.1.4.2: (for components mainly subjected to bending)

- x/d ≤ 0.35: Internal force redistributions **without** verification of deformation capacity
- 0.35 ≤ x/d ≤ 0.5: Internal force redistributions **with** verification of deformation capacity
-

Limitation of the compression zone depth according to SIA 262

System behaviour (see also [6], page 2.32ff)

Continuous increase of the load *q*:

- \rightarrow Yielding begins first at the fixed-end support, forming a plastic hinge
- \rightarrow The statically indeterminate system turns (for additional loading) into a simple beam

Further load increase is possible until a second plastic hinge is formed in the member $(=$ mechanism $):$

- \rightarrow Plastic rotation required at the fixed support
- → **Rotation demand** depending on static system and load configuration
- \rightarrow **Rotation capacity** limited by steel elongation and / or concrete compression

Verification = Comparison: Deformation capacity Θ_{p} _{*pu*} ↔ Deformation demand Θ_{p} _{*pu,dem*}

Rotation demand $\Theta_{p\mu, dem}$ (approximation, example two-span beam) In general, deformation capacity and deformation demand are coupled. The interaction can only be neglected for moderate redistributions. Additional simplifications: **Bea**
approximation, exacity and deformation, exactly and deformation
is associal and $\Theta_{pu,dem}$ of the rotation of the two can be considered and $q - q_y/l^3$
 $\frac{q - q_y/l^3}{12EI}$

- Constant bending stiffness
- *M-* Θ rigid-ideal plastic (no hardening in the plastic hinge)

Therefore, the rotation demand $\Theta_{p\mu, dem}$ of the intermediate support corresponds to the relative rotation of the two beams over the intermediate support, which can be considered as simply supported beams after reaching *May* (at $q = q_y$): only be neglected
 ations:
 *p*g stiffness

plastic (no hardenir

tion demand $\Theta_{pu,de}$

relative rotation of

ort, which can be comp M_{ay}
 $\log M_{ay} = \frac{\left(q - q_y\right)l^3}{12EI}$

$$
\Theta_{pu, dem} = \frac{(q - q_y)l^3}{12EI}
$$

(Two-span beam, first plastic hinge at intermediate support, deformation demand for full load)

Rotation demand - Example of a two-span beam

Rotation demand - Example of a two-span beam

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Rotation capacity Q*pu* **– Basics**

Example: Plastic hinge angle as a function of ω (ductility classes A-C, 1999)

Basis of the calculations:

Rotation capacity $\Theta_{\rho\nu}$ (simplified) (see also [6], page 2.32ff)

Limitation of the plastic rotation by the reinforcing steel (rupture of the reinforcement):

> $\left(\sqrt{\varepsilon}_{\textit{smu}}\right)$ $\left(\sqrt{\varepsilon}_{\textit{smv}}\right)$ Curvature at $\Theta_{\textit{\tiny{pus}}} = L_{\textit{\tiny{pl}}}\cdot\left(\left(\frac{\sigma_{\textit{\tiny{smul}}}}{d-x}\right)^{\frac{1}{2}}\left(\frac{\sigma_{\textit{\tiny{smul}}}}{d-x}\right)^{\frac{1}{2}}\right)$ Curvature at rup

 $d-x$ $d-x$ *)* Curvature at rupture of the Curvature at onset of yielding reinforcement

Limitation of the plastic rotation by the concrete (compressive failure):

> $p_{\mu c} = L_{pl} \left[\begin{array}{c} \frac{-c_2 a}{\sigma} \end{array} \left| \begin{array}{c} \frac{s m y}{\sigma} \end{array} \right| \right]$ $\Theta_{puc} = L_{pl} \left(\begin{array}{c} \frac{c_{c2d}}{x} \\ x \end{array} \right) \left(\frac{smy}{d-x} \right)$ Curvature at concrete

 $\left\langle \left\langle \varepsilon_{c2d}^{\mathrm{max}}\right\rangle \right\rangle$ *Curvature at onset of yielding* crushing

Rotation per crack: $\Theta_i \approx \frac{\mathcal{S} \textit{sm}^{\omega} \textit{rm}}{\mathcal{S} \textit{mm}}$ *i s* $d - x$ $\Theta_{\cdot} \approx \frac{\mathcal{E}_{sm} S_{rm}}{\mathcal{E}_{\cdot}}$ −

Plastic hinge rotation = sum of the plastic rotations of all cracks from the onset of yielding

- $_{pl}$ $\,$ Plastic hinge length, depending on load configuration and geometry: region in which the chord reinforcement yields $(\rightarrow$ determine the chord force distribution from the stress field). L_{nl}
- Mean steel elongation when reaching *smu*
- Mean steel elongation when reaching *smy*

$$
\varepsilon_{sr} = \varepsilon_{ud} \qquad \sigma_{sr} = f_t \qquad \varepsilon_{sr} \leftrightarrow \varepsilon_{sm}
$$

$$
\varepsilon_{sr} = \frac{f_s}{E_s} \qquad \sigma_{sr} = f_s \qquad \text{(Stahlbeton I)}
$$

sr

Rotation demand Rotation capacity (simplified) - Example of a two-span beam

$$
\Theta_{puc} = L_{pl} \cdot \left(\frac{\varepsilon_{cu}}{x} - \frac{\varepsilon_{smy}}{d^--x}\right)
$$
\nwith $\frac{\varepsilon_{smy}}{d^--x}$ = Curvature at onset of yielding $= \frac{f_s/E_s - \Delta \varepsilon_0}{d^--x} = 2.3$ mrad/m, L_{pl} = length plastic hinge $\approx 2d^-$
\n $\Theta_{pus} = L_{pl} \cdot \left(\frac{\varepsilon_{smu}}{d^--x} - \frac{\varepsilon_{smy}}{d^--x}\right)$

Rotation demand Rotation capacity (simplified) - Example of a two-span beam

Rotation at failure:

Concrete crushing

$$
\Theta_{puc} = L_{pl} \cdot \left(\frac{\varepsilon_{cu}}{x} - \frac{\varepsilon_{smy}}{d^- - x}\right) \approx 2 \cdot 1.10 \cdot \left(\frac{0.003}{0.181} - 0.0023\right) = 14.3 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 31.4 \text{ mrad}
$$

\n
$$
\rightarrow \Theta_{puc} > \Theta_{B,dem} \rightarrow \text{OK}
$$

Steel rupture

rough assumption: $\varepsilon_{\rm s_{mu}} \approx 0.5 \varepsilon_{\rm u} = \frac{1}{2} \varepsilon_{\rm e} \approx 0.5 \varepsilon_{\rm e}$ (estimated reduction of elongation at failure 22.5‰ (B500B) | 22.5‰ (B500B) | $\mathcal{L}_{smu} \approx 0.5 \varepsilon_{ud} = \begin{cases} 32.5\% & \text{(B500C)} \end{cases}$ (estimated red $(22.5\% \text{ (B500B)})$ $\varepsilon_{\textit{smu}} \approx 0.5 \varepsilon_{\textit{ud}} = \begin{cases} 22.5\% & (\text{B500C}) \\ 32.5\% & \text{(B500C)} \end{cases}$ (estimately $(B500B)$ and (A, B, B) 50

(B500C) due to tension stiffening - see next slides)

Beams − **Deformation capacity**
\nstation at failure:
\n
$$
\Theta_{\text{pw}} = L_{\mu} \cdot \left(\frac{\varepsilon_{\text{w}}}{x} - \frac{\varepsilon_{\text{w}}}{d^2 - x} \right) \approx 2 \cdot 1.10 \cdot \left(\frac{0.003}{0.181} - 0.0023 \right) = 14.3 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 31.4 \text{ mrad}
$$
\n
$$
\Theta_{\text{pw}} \approx \Theta_{\text{p},\text{a}} \Rightarrow \Theta_{\text{p},\text{a}} \rightarrow \text{OK}
$$
\n
$$
\Theta_{\text{p}} \approx \Theta_{\text{p},\text{a}} \rightarrow \text{OK}
$$
\n
$$
\Theta_{\text{p}} \approx \Theta_{\text{p},\text{a}} \rightarrow \text{OK}
$$
\n
$$
\Theta_{\text{p}} = L_{\mu} \cdot \left(\frac{\varepsilon_{\text{p}}}{d^2 - x} - \frac{\varepsilon_{\text{p}}}{d^2 - x} \right) = \begin{cases} 22.5\%_{\text{m}} \left(\text{B500B} \right) \\ 32.5\%_{\text{m}} \left(\text{B500C} \right) \end{cases} \quad \text{(estimated reduction of elongation at failure due to tension siffening - see next slides)}
$$
\n
$$
\Theta_{\text{p},\text{a}} = L_{\mu} \cdot \left(\frac{\varepsilon_{\text{p}}}{d^2 - x} - \frac{\varepsilon_{\text{p}}}{d^2 - x} \right) = \begin{cases} 2 \cdot 1.10 \cdot \left(\frac{0.0225}{0.919} - 0.0023 \right) = 22.2 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 48.8 \text{ mrad (B500B)} \\ 2 \cdot 1.10 \cdot \left(\frac{0.0325}{0.919} - 0.0023 \right) = 33.1 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 72.7 \text{ mrad (B500C)} \end{cases}
$$
\n
$$
\Rightarrow \Theta_{\text{p},\text{a}} \approx \Theta_{\text{p},\text{a}}
$$

The rotation capacity would be verified. **But**: Are the assumptions of L_{ph} , ε_{smu} all right?

Rotation capacity (detailed investigation) - Example of a two-span beam

1 2.2% $\frac{df}{dt} = \frac{M_r(d^--x)E_s}{m} + 1 - n$ $\rho_{\text{eff}} = \frac{1}{M_r (d^2 - x) E_{s+1}} = 2.2\%$ *II* $ct - 1$ *n* f_{ct} **EI**^{μ} $+$ n • C30/37: $f_{cd} = 20 \text{ MPa}, f_{ctm} = 2.9 \text{ MPa}$ • $d^- \approx 1.1$ m, A_s $f_{sd} = 1848$ kN Equivalent reinforcement ratio (considering *x* at failure, see notes *): $A_s = 4240 \text{ mm}^2$

0.8 $d' - x$

0.60 $\overrightarrow{A_s} = 4240$ mm² *sm* d' – x *x* 1 $\frac{1848}{181}$ = 181 mm $x = \frac{0.85 \cdot 0.6 \cdot 20}{0.85 \cdot 0.6 \cdot 20} = 181$ mm d^{-} – x = 919 mm $\approx 1.1 \text{ m}, A_s$ $f_{sd} = 1848 \text{ KN}$
 $x = \frac{1848}{0.85 \cdot 0.6 \cdot 20} = 181 \text{ mm}$ \rightarrow x = \rightarrow = \rightarrow = \rightarrow 181 mm \cdot \cup . \cup \cdot \angle \cup

 $\alpha_0 \approx \frac{\varnothing}{\cdot} \left(\frac{1}{1} - 1 \right) = 292$ mm $\left(\lambda = \frac{1}{2} \ldots 1 \right)$ $S_{rm0} \approx \frac{1}{4} \cdot \left(\frac{1}{8} - 1 \right) = 292 \text{ mm } \left(\lambda = \frac{1}{2} \cdot \lambda \right)$ *t* $\approx \frac{\varnothing}{4} \cdot \left(\frac{1}{\rho_t} - 1\right) = 292$ mm $\left(\lambda = \frac{1}{2} ... 1\right)$

 \rightarrow $s_{\scriptscriptstyle rm}$ \approx 250 mm (spacing of stirrups)

Rotation capacity (detailed investigation) - Example of a two-span beam Tension chord model $\tau_{b0} = 2 f_{cm} = 5.8 \text{ MPa}$ $\tau_{b1} = 1 f_{cm} = 2.9 \text{ MPa}$ \varnothing = 26 mm $s_{rm}=250\;\mathrm{mm}$ (spacing of stirrups)
 $E_{\parallel}=205\;\mathrm{GPa}$ $= 205$ GPa $f_{\rm c\it m} = 2.9 \, \rm MPa$ 1 nackter Stahl $\Delta \varepsilon_0 = \Delta \varepsilon^{[\underline{1}]}$ 0 0.27‰ *sr b r sr* $2\tau_{b1}S_r$ = 56Mpc $2\hskip-3.5pt\relax o_{\mathit{sr}}^{} > f_{\mathit{s}}, \ \sigma_{\mathit{smin}}^{} < f_{\mathit{s}}^{} \ \rightarrow ...;$ Transition to regime $\hskip-3.5pt\relax o_{\mathit{st}}^{}$ at $b1^{3}r = \sigma$ -56 MPa $\stackrel{\text{!}}{=}$ f 1 $\sigma_{\rm\scriptscriptstyle SF}$ $<$ $f_{\rm\scriptscriptstyle s}$ "elastic" $\frac{3}{5}$ $\sigma_{\text{spin}} > f_s$ $\rightarrow \varepsilon_{\text{sm}} = \frac{J_s}{F} + \frac{S_s - S_s}{F} - \frac{bS_r}{F}$ $\rightarrow \sigma_{sr}^{} = f_s^{} + 56~\mathrm{MPa} \rightarrow \mathsf{B}500\mathsf{B}$ stays at regime $[2] \qquad \vdots$ *sm* \blacksquare $s = s²$ s s $smin$ S_{sr} \sim S_{sr} so that a J_{s} $f_s \longrightarrow \varepsilon_{sm} = \frac{f_s}{E_s} + \frac{\sigma_{sr} - f_s}{E_{sh}} - \frac{\tau_{b1} s_r}{E_{sh} \varnothing}$ *s E E E* $\frac{\tau_{b1}S_r}{\sim} = \sigma_{sr} - 56 \text{ MPa} \stackrel{!}{=} f_s$ $\sigma_{sr} < f_s$ "elastic"
 $\rightarrow \varepsilon_{sm} = \frac{\sigma_{sr}}{E_s} - \frac{\tau_{b0} s_r}{E_s \varnothing} = \frac{\sigma_{sr}}{E_s} - 0.27\%$ o

nackter Stahl
 $\Delta \varepsilon_0 = \Delta \varepsilon^{[1]}$ $\sigma_{\textit{smin}} = \sigma_{\textit{sr}} - \frac{\sigma_{\textit{sr}}}{\varnothing} = \sigma_{\textit{sr}} - 50 \text{ MPa} = f_{\textit{s}}$ $\sigma_{smin} > f_s$ $\longrightarrow \varepsilon_{sm} = \frac{f_s}{\underline{E}_s} + \frac{\sigma_{sr} - f_s}{\underline{E}_{sh}} - \frac{\tau_{b1} s_r}{\underline{E}_{sh} \oslash \frac{\tau_{b1} s_{ch}}{\underline{E}_{sh} \oslash \frac{\tau_{b2} s_{ch}}{\underline{E}_{sh} \oslash \frac{\tau_{b3} s_{ch}}{\underline{E}_{sh} \oslash \frac{\tau_{b4} s_{ch}}{\underline{E}_{sh} \oslash \frac{\tau_{b5} s_{ch}}{\underline{E}_{sh} \oslash \frac{\tau_{b4} s_{ch}}{\underline{E}_{sh} \oslash \frac{\tau_{b5} s_{ch}}$ 3 $1 - \Delta G$ "partially yielded" ! $\frac{1}{1}$ Dovoo $3\frac{1}{2}$ $\frac{1}{2}$ $\frac{1}{2}$ 2 \sim \sim \sim \sim B500C 30‰ (B500B): 2 3‰ (B5 00C) 0‰ (B500B): % (В500В); в соборах производство в соборах производство в соборах производство в соборах производство в собор (B500B): with a set of \overline{B} 00B): with a set of \sim $1^{\prime\prime}$, $\frac{1}{\cdot}$ $29MPa$ $s \leftarrow b^2$ $\sigma_{sr} = f_{sd} + E_{sh} \left(\varepsilon_m - \frac{f_s}{E_s} \right) + \frac{\tau_{b1} s_r}{\varnothing}$ *s MPa E* (f) τ_{ij} s **B500C:** $\varepsilon_{sm}(\sigma_{sr} = f_s) = 2.43 - 0.27 = 2.16%$ $\varepsilon_{_{SM}} (\sigma_{_{Smin}} = f_{_S}) = 25.9\%$ (3 with $\sigma_{_{SF}} = 556 \text{ MPa}$) $\varepsilon_{_{SM}}$ $(\sigma_{_{SF}} = f_t) = 65 - 23 = 42\%$ $\sigma = \varepsilon_{_{SMU}}$ **B500B**: $\varepsilon_{_{SM}}(\sigma_{_{SF}} = f_{_S}) = 2.16\%$ $\epsilon_{sm}(\sigma_{sr} = f_t) = 0$ 3 – 23 = 42% $\sigma = \epsilon_{smu}$
 $\epsilon_{sm}(\sigma_{sr} = f_s) = 2.16\%$
 $\epsilon_{smu} = 17.7\%$ (Regime 2 with $\sigma_{sr} = f_t$, does not reach regime 3)

Rotation capacity (detailed investigation) - Example of a two-span beam

Rotation capacity (detailed investigation) - Example of a two-span beam

Rotation capacity (detailed investigation) - Example of a two-span beam Plastic hinge length → Distribution of the top chord force *Fsup* determined from a stress field

Rotation capacity (detailed investigation) - Example of a two-span beam Plastic hinge length \rightarrow Distribution of the top chord force F_{sub} determined from a stress field

 R_{Bd} increases during the redistribution, x_P thus decreases (large gradient of M is unfavourable for the rotational capacity, since a stronger localization of deformations occurs):

 R_{Bd} (and thus x_p) also depends on the choice of the compression field inclination α_0 :

 $\overline{0}$ large $\alpha_{_0} \rightarrow x_{\rm p_1}, x_{\rm p_2}$ small, small $\alpha_{_0} \rightarrow x_{\rm p_1}, x_{\rm p_2}$ large $2z \cot(\alpha_0)(\overline{q}_d + f_{wd}) \rightarrow (\overline{q}_d + f_{wd}) = \frac{R_{Bd}}{2z \cot(\alpha_0)}$
 $\alpha_0 \rightarrow x_{B1}, x_{B2}$ small, small $\alpha_0 \rightarrow x_{B1}, x_{B2}$ large $Bd \sim 2.$ $\cot(\omega_0/\sqrt{d} + J_{wd})$ $\wedge \sqrt{d} + J_{wd}$ *R* $R_{Bd} = 2z \cot(\alpha_0)(\overline{q}_d + f_{wd}) \rightarrow (\overline{q}_d + f_{wd}) = \frac{f_{wd}}{2}$ z COU α _o) $= 2z \cot(\alpha_0)(q_x + t_{y} + t_{y} + \epsilon) \Rightarrow (q_x + t_{y} + t_{z} + t_{z} + \epsilon) =$ α_{α})

- \rightarrow Several assumptions are necessary to determine the deformation capacity
- \rightarrow Rough estimation, not exact calculation!

 $\frac{1}{2}$ 2

Rotation capacity (detailed investigation) - Example of a two-span beam

Plastic hinge length → Distribution of the top chord force *Fsup* determined from a stress field

$$
\rightarrow \text{Assumption:} \qquad R_{\text{Bd}} \approx 1500 \text{ kN}, \ \cot(\alpha_0) = 1.5 \ (\alpha_0 = 33.5^\circ), \ \left(\overline{q}_d + f_{\text{wd}}\right) = \frac{1500}{2 \cot(\alpha_0)} = \frac{1500}{3} = 500 \text{ kNm}^{-1}
$$

B500B :
$$
x_{P2} = \sqrt{\frac{2 \cdot 4240(540 - 500) \cdot 1000}{500}} = 823 \text{ mm } (\degree L_{p_1}/2^{\degree})
$$

\nB500C : $x_{P2} = \sqrt{\frac{2 \cdot 4240(575 - 500) \cdot 1000}{500}} = 1127 \text{ mm } (\degree L_{p_1}/2^{\degree})$
\n $x_{P1} = \sqrt{\frac{2 \cdot 4240(575 - 556) \cdot 1000}{500}} = 571 \text{ mm}$
\n6 $\text{gauge} [\%o]$
\n**40**
\n**41**
\n**6**
\

Rotation demand and rotation capacity (detailed investigation) - Example two-span beam Plastic rotation at failure

Concrete crushing

$$
\Theta_{puc} = L_{pl} \cdot \left(\frac{\varepsilon_{cu}}{x} - \frac{\varepsilon_{smy}}{d^- - x}\right) \approx 2.1.10 \cdot \left(\frac{0.003}{0.181} - 0.0023\right) = 14.3 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 31.4 \text{ mrad}
$$

\n
$$
\rightarrow \Theta_{puc} > \Theta_{B,req} \rightarrow \text{OK}
$$

Steel rupture

0.5 32.5‰ 2.2 m *pl smu ud pl L L* B500 B500 B C Rough assumption: *smu* B500B B500C More detailed investigation:

Beans – **Deformation capacity**
\nand **rotation capacity (detailed investigation)** – **Example two-span beam**
\n
$$
\Theta_{\text{par}} = L_{\text{par}} \cdot \left(\frac{\varepsilon_{\text{var}}}{x} - \frac{\varepsilon_{\text{amp}}}{d - x} \right) \approx 2 \cdot 1.10 \cdot \left(\frac{0.003}{0.181} - 0.0023 \right) = 14.3 \frac{\text{mrad}}{\text{m}} \cdot 2.2 \text{ m} = 31.4 \text{ mrad}
$$
\n
$$
\rightarrow \Theta_{\text{par}} > \Theta_{\text{Bar}} \rightarrow \text{OK}
$$
\nRough assumption:
$$
\varepsilon_{\text{max}} \approx 0.5 \varepsilon_{\text{ad}} = \left\{ \frac{22.5\% \text{ with } L_{\text{at}} = 2.2 \text{ m (B500E)} \right\}
$$
\nMore detailed investigation:
$$
\overline{\varepsilon}_{\text{max}} = \left\{ \frac{10.5\% \text{ with } L_{\text{at}} = 1.65 \text{ m (B500C)} \right\} \left[\overline{\varepsilon}_{\text{amp}} \approx 0.23 \cdot \varepsilon_{\text{at}}, L_{\text{at}} = 1.5 \cdot d \right]
$$
\n
$$
\Theta_{\text{par}} = L_{\text{par}} \cdot \left(\frac{\overline{\varepsilon}_{\text{amp}}}{d^2 - x} - \frac{\overline{\varepsilon}_{\text{amp}}}{d^2 - x} \right) = \left\{ \frac{1.65 \cdot \left(\frac{0.0105}{0.919} - 0.0023 \right) \right\} = 15.1 \text{ mrad (B500E)} \times \Theta_{\text{amp}} = 18.5 \text{ mrad} (\alpha_{\text{at}} = 1)}{\text{not fulfilled}} \right\}
$$
\n
$$
\Theta_{\text{par}} = L_{\text{par}} \cdot \left(\frac{\overline{\varepsilon}_{\text{amp}}}{d^2 - x} - \frac{\overline{\varepsilon}_{\text{amp}}}{d^2 - x} \right) = \left\{ \frac{1.65 \cdot \left(\frac{0.0105}{0.919} - 0.0023 \right) \right\} = 53.8 \text{ mrad (B500C)}
$$

Additional considerations: ratio of mean strain to maximum strain in the cracks considering bond

[Alvarez 1999]

Additional considerations: influence of the reinforcement hardening properties

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Additional considerations: influence of the reinforcement hardening properties

- \rightarrow Reinforcement with a yield plateau is more favourable than cold-formed reinforcement, especially in case of failure in regime $\boxed{2}$ (yield plateau contributes as an "additional" strain over the entire yielded area)
- \rightarrow The bilinear idealization overestimates the deformation capacity for a reinforcement with high ductility

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Tension experiments – Dr. M. Alvarez: Test setup

- **1** Stahl-Reaktionsrahmen **3** Druckstreben (Knicksicherung) 8 Verankerungen der Anschlussstäbe (3) Servohydraulische Pressen (2 MN Kapazität) **9** Verankerung des Vorspannkabels 4 Oberer Krafteinleitungsträger ^{(@} Vorspannstangen (Verankerungen im Aufspannboden) **5** Versuchskörper ⁽¹⁾ Kraftmessdosen (1 MN Nennbereich) **3** Kraftmessdose (2 MN Nennbereich)
- **6** Unterer Krafteinleitungsträger

3000 1000

Bild 1.1 - Konzept der Zugversuche, [mm].

Tension experiments – Dr. M. Alvarez: Crack patterns at failure

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Tension experiments – Dr. M. Alvarez: Test results

Load-deformation behaviour considering bond

 \rightarrow Deformation capacity severely impaired for reinforcement with low ductility (failure deformation and hardening!)

Ratio of average elongation to maximum elongation at the cracks considering bond

 \rightarrow Good agreement with tension chord model (almost identical if the real bare steel curve is taken into account)

Deformation capacity

Summary

- The concrete strength should be reduced in plastic analysis depending on the cracking state of the structure and on the material brittleness.
- The concrete contribution in tension between two cracks stiffens the response of bonded reinforcement with respect to bare (unbonded) reinforcement. This tension-stiffening effect affects the serviceability response of the structure but also reduces the deformation capacity of the reinforcement. Assuming simplified bond relationships (as e.g. in the Tension Chord Model) is sufficient for modelling tension-stiffening.
- *Deformation capacity* and *deformation demand* are coupled. The interaction can only be neglected for moderate redistributions of the internal forces.
- The *deformation demand* can be determined approximately with reasonable effort using simplified assumptions (constant bending stiffness of the elastic areas, rigid-ideal plastic *M-*Q *relationships of* the plastic hinges).
- Even with complex calculations, the *deformation capacity* can only be roughly estimated because it depends on several effects and assumptions that cannot be precisely quantified:
	- − Bond behaviour, in particular, crack spacing
	- − Mechanical properties of the reinforcement (hardening ratio and deformation of failure, with or without yield plateau)
	- − Force flow in the area of plastic hinges, in particular, variation of the force in the tension chord \rightarrow the mean deformations averaged over the length of the plastic hinge are smaller than the mean deformation of a tension chord under constant tensile force!)
- In practice, it is therefore advisable to avoid the verification of the deformation capacity for new structures whenever possible (complying with the condition x/d < 0.35). Otherwise, it is often easier to ignore the redistribution of internal forces, i.e. to verify the structural safety for the elastic stresses including restraint stresses (even if the estimation of the restraints is also time-consuming and requires assumptions).
- If the deformation capacity needs to be verified (e.g. for existing structures), engineering judgement must be applied. The decisive parameters should be accounted for as accurately as possible (reinforcement: determine hardening characteristics, not just *f^s*).